

# Time & position Calibration with known sources

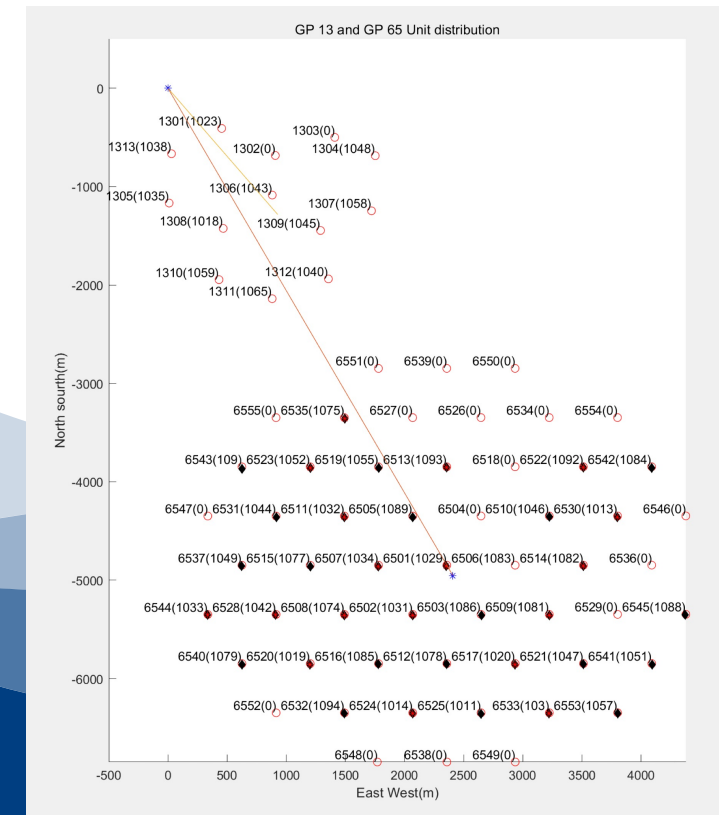
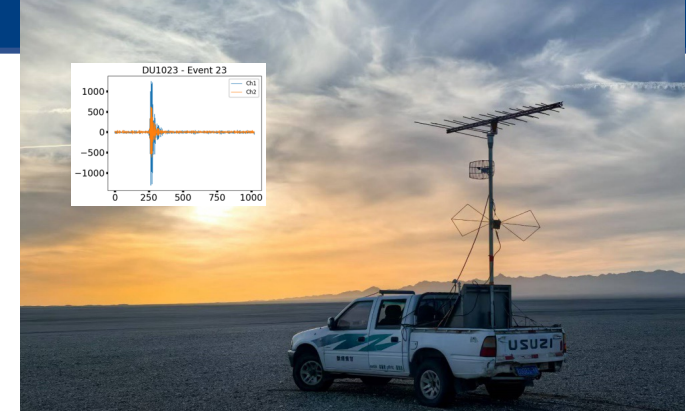
Dec 05<sup>th</sup> 2024 (Xidian, Pengfei Zhang)



- Xidian
- International Scientific Event “GRAND” Collaboration Meeting, 2025, Warsaw Poland

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- (1) Calibration Background:
  - Time accuracy requirement
- (2) Beacon base calibration method (First site experiment)
  - Hardware realization
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  - Hardware
  - Data processing



# Time Calibration Background

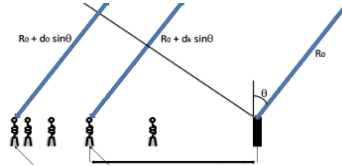
## (1) For Incident direction reconstruction

$$\begin{cases} S_1(t) = A_1 \cos(\omega t_1) \\ S_2(t) = A_2 \cos(\omega(t_1 + \Delta t)) \end{cases}$$

$$\Delta t = d \sin \theta_i / c$$

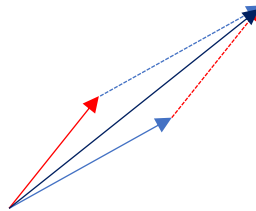
$$\theta_i = \arcsin(\Delta t c / d)$$

$$\left| \frac{d\theta_i}{d\Delta t} \right| = \left| \frac{1}{\sqrt{1 - \sin^2(\theta_i)}} \frac{c}{d} \right| = |c / d \cos(\theta_i)|$$

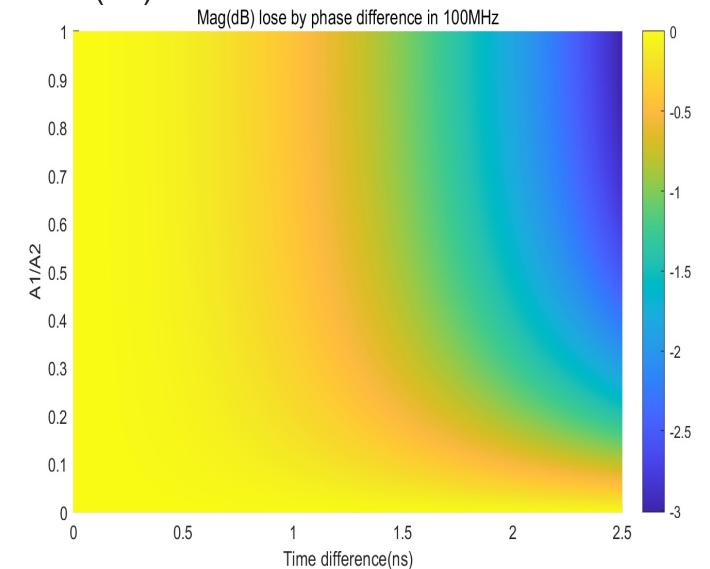
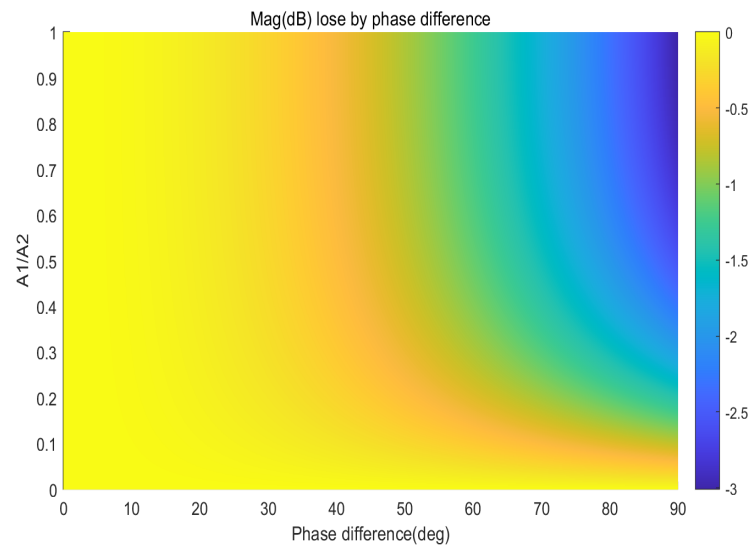
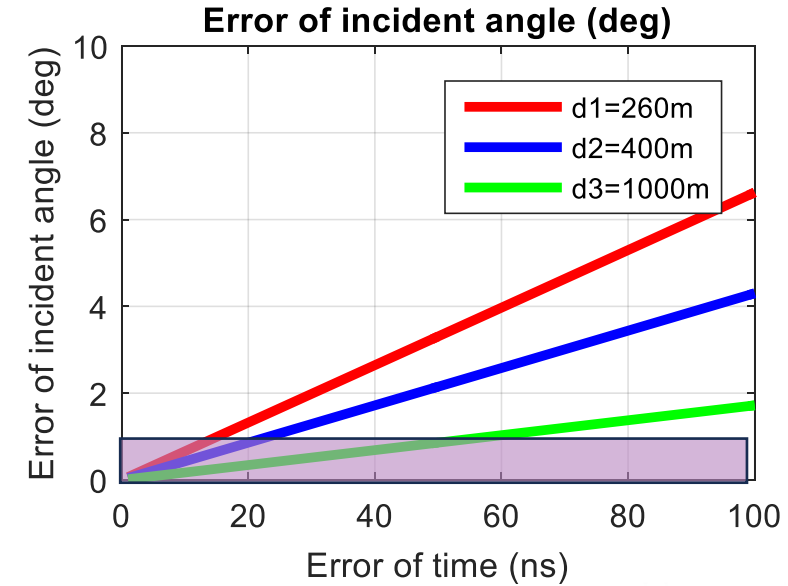


## (2) For interferometric observation

$$\begin{cases} S_1(t) = A_1 \cos(\omega t + \varphi_1 + \sigma_{12}(t)) \\ S_2(t) = A_2 \cos(\omega t + \varphi_2) \end{cases}$$



$$|S_1(t) + S_2(t)| / |A_1 + A_2| = (A_1^2 + A_2^2 + 2A_1A_2 \cos(\sigma_{12}(t)))^{1/2} / |A_1 + A_2|$$



# Time Precision requirement

- **Ref:Nanosecond-level time synchronization of autonomous radio detector stations for extensive air showers:**

“For ns-timing precision, a maximum phase error of  $20^\circ$  can be tolerated. The  $20^\circ$  correspond to a shift of slightly less than 1ns, since a full period ( $b= 360^\circ$ ) is about 17ns for the lowest beacon frequency.”

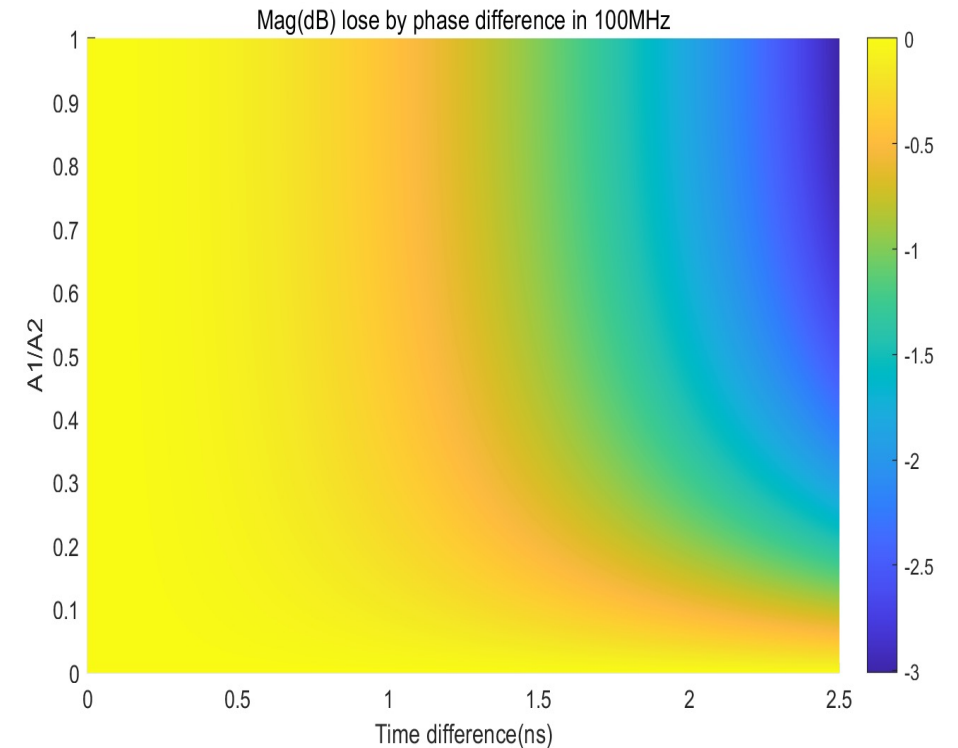
- **Antenna hand book**

The maximum phase error within the aperture should less than  $22.5^\circ$  which means that the time precision should less than 1/16 of the full period . For 200MHz and 5ns period, this lead to a very challenge time precision of **0.3125ns**.

**Hence a rough requirement is 2ns**

**restrict requirement is 0.2~0.5ns**

It should be noted that in some cased the **time precision requirement can be relaxed while pay the cost of long Coherent integration observation time.**





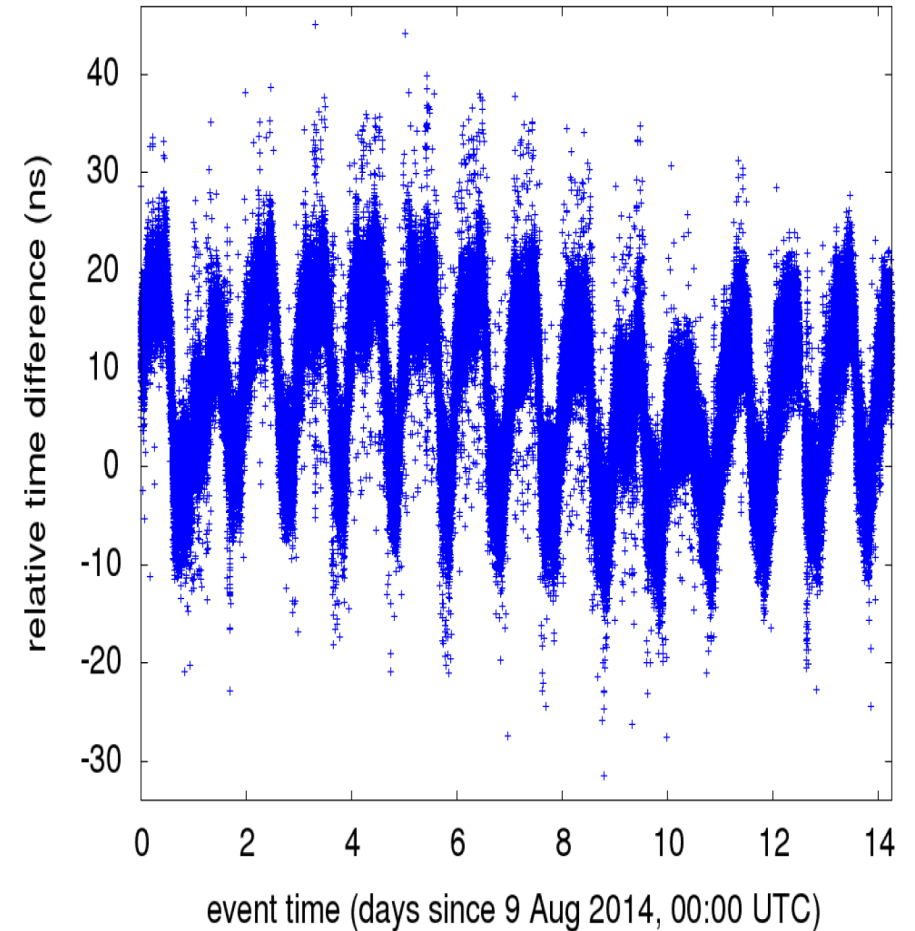
# How to describe the error in GPS based Time synchronization system

**A :** The GPS time system it self have three kinds of error which have been verifier by the related work done in AERA.

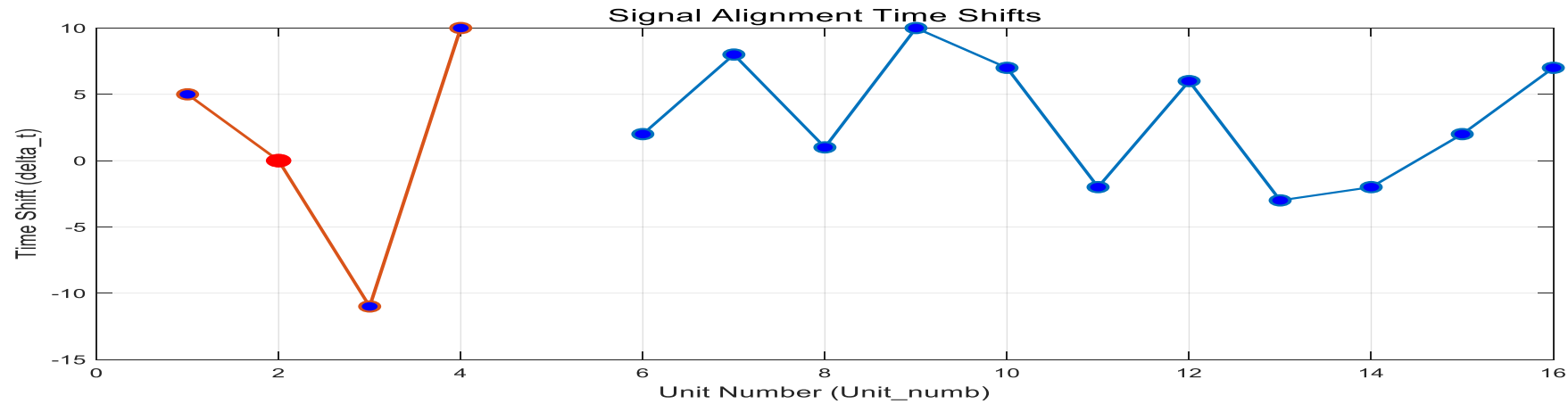
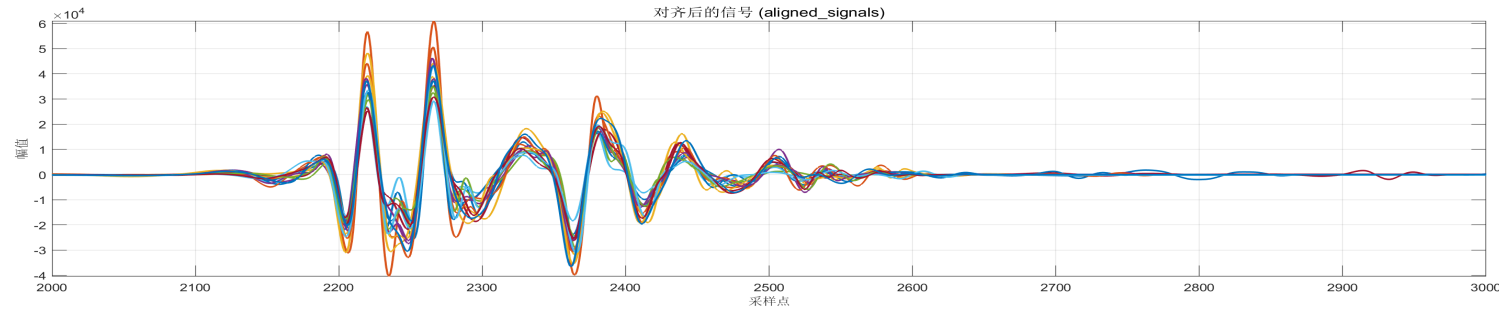
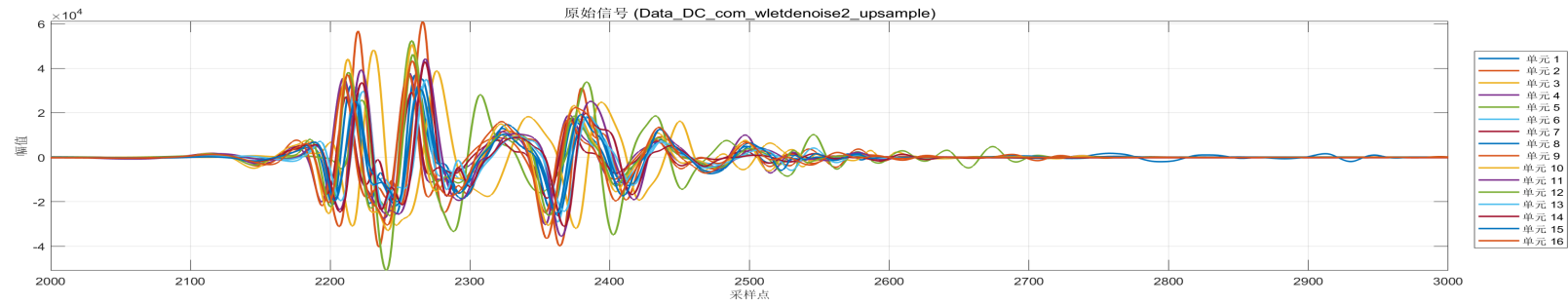
- It include the following parts:

$$\sigma_{GPS} = \sigma_{GPS\_SR}(t) + \sigma_{GPS\_LF}(t) + \sigma_{GPS\_Feb}(n)$$

- $\sigma_{GPS\_SR}(t)$  present for short-term, rapid random error on the nanosecond scale, with a maximum variation of approximately 10 ns and a standard deviation (STD) of 5 ns.
- $\sigma_{GPS\_LF}(t)$  present for the long-term fluctuation spanning several hours or even a day, with fluctuation amplitudes reaching 30-50 ns.
- The first one is acceptable for direction reconstruction for sparse array and large incident angle with  $0.2^\circ$  precision. While both of them should be compensated by calibration for interferometric observation



# Trigger algorithm based time shift will not take into consideration any more (as mentioned yesterday it can be solved based on wavelet and correlation ship )

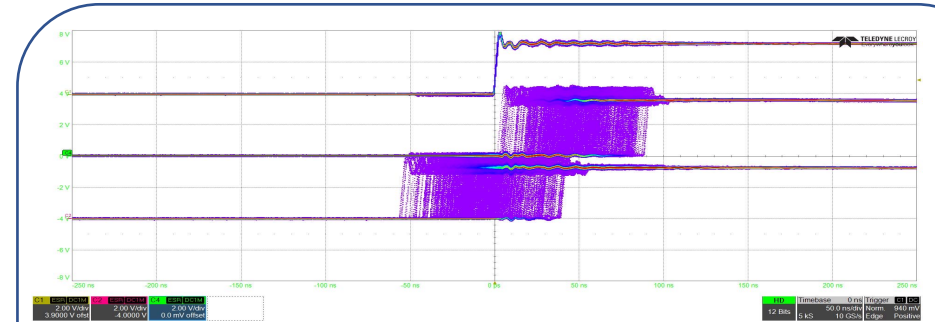


## How to describe the Time error in GPS based system

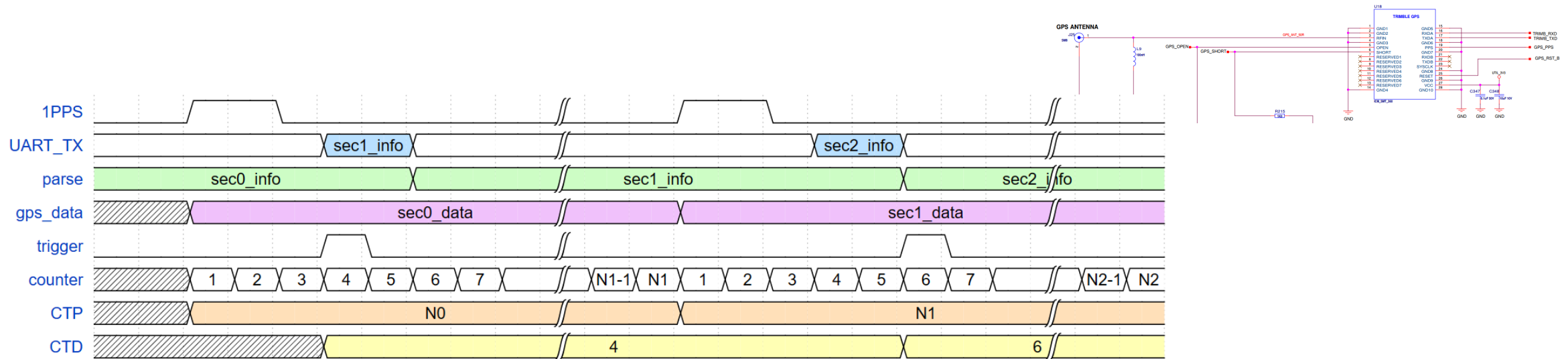
The Feb hardware system cause shift of Units time.

$$\sigma_{GPS \text{ Feb}}(n)$$

- $\sigma_{Feb}(n)$  present for the stable system shift of GPS time system (which have not be reported in AERA) and for GRAND. It is firstly be observed by Xing xu within laboratory test.



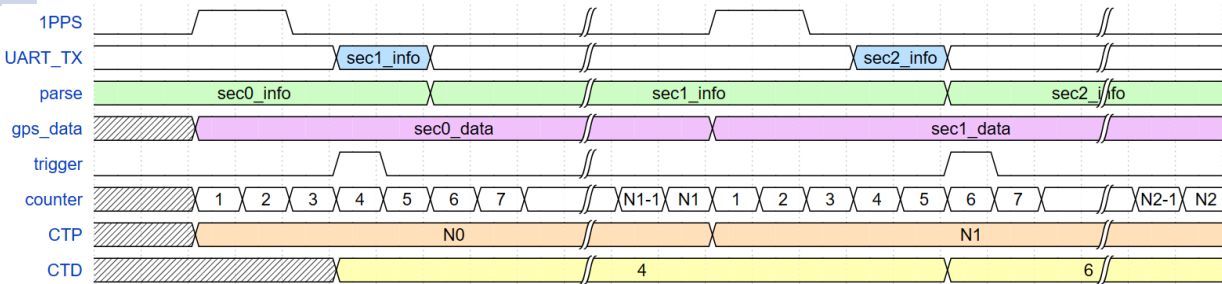
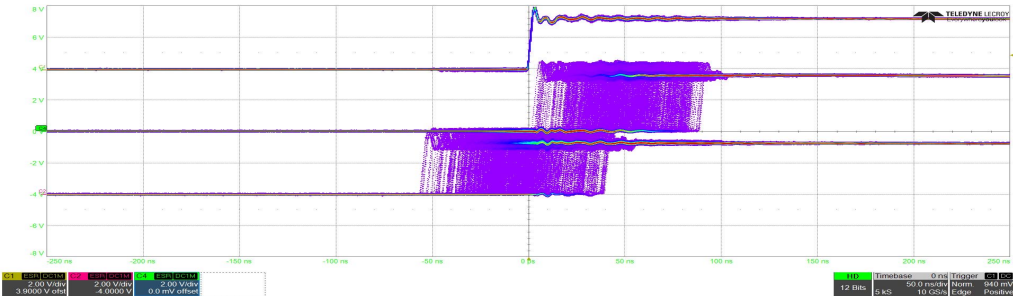
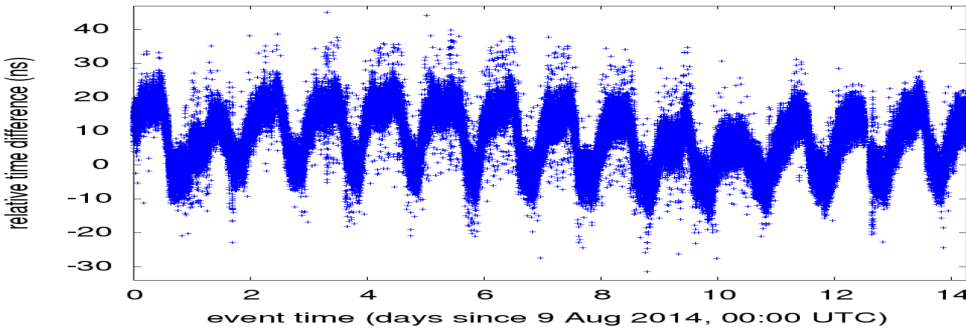
Xing Xu2024.04.01 HW report



# Time error in GPS based system

$$\sigma_{GPS} = \sigma_{GPS\_SR}(t) + \sigma_{GPS\_LF}(t) + \sigma_{GPS\_Feb}(n)$$

	Typical value(ns)	time-dependent features	Incident angle-dependent
$\sigma_{GPS\_SR}$	5~10	Short time random	No
$\sigma_{GPS\_LS}$	0~50	Long time Fluctuation	No
$\sigma_{GPS\_Feb}$	0~50	Stable	No





# Beacon based calibration and it's limitation

Hardware (Ref AERA) :

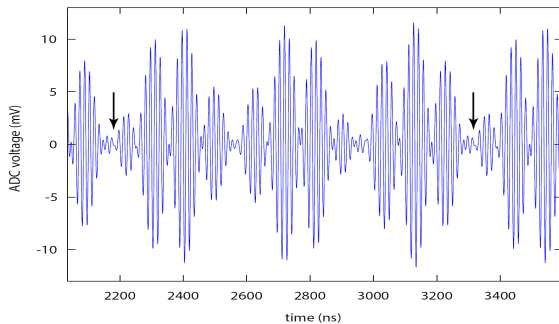
- Signal generator
- Attenuators, filters and amplifiers
- Transmitter antenna

Frequency choice:

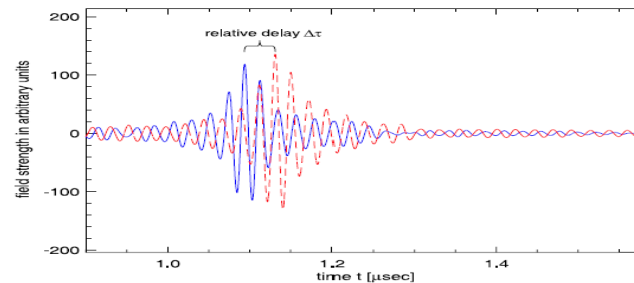
- Time calibration precision/sample rate/Antenna beam pattern;
- High order suppress/Environment interfere

Note:

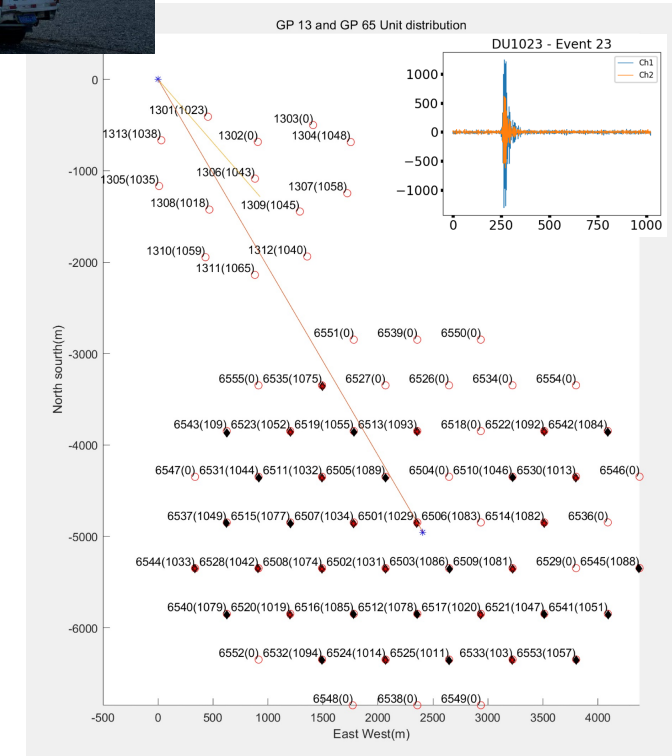
- Sample frequency;(500MHz)
- Record length; (8192)
- Port choice ; (Y/Z)
- Mixing and high order suppress (-40dB)
- Arriving time define and acquisition algorithm



Nanosecond-level time synchronization of autonomous radio detector stations for extensive air showers



New method for the time calibration of an interferometric radio antenna array

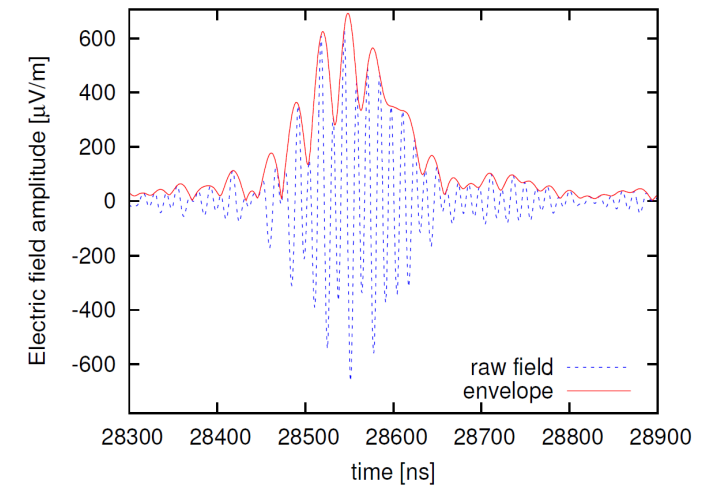
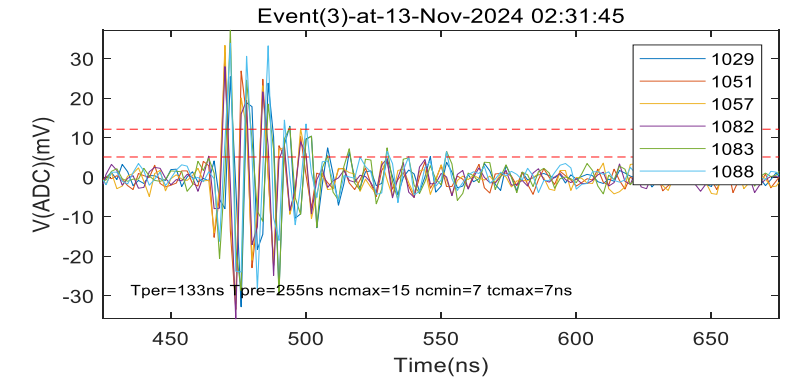


# Beacon based TR process-- relative time delay

## A : The time delay include:

$$T_{TR\_beacon} = T_{T\_transmitter} + T_{P\_process} + T_{R\_antenna} + T_{R\_RF\_Chain} + T_{Feb\_sample}$$

- $T_{T\_transmitter}$  present for the transmitter process, what we should pay attention is the distortion caused by dispersion, which may lead error in arriving time determine.
- $T_{P\_process} = l_n / (c / n_{air})$  present for the Radio wave travel time, The problem is that we should confirm that the whether radio waves travel as free space plan wave at the speed of light in the atmosphere or possibly as surface of ground as crawl waves. There is a big difference between the time delay between them. The test results now available show that the former one is correct, but if we using the Z channel , we should check it again. Another problem is the precision of  $n_{air}$  will lead problem in long distance test such as 10km.



# Beacon based TR relative time delay

- $T_{R\_antenna}$  present for the Respond time of antenna, the problem is that if pulse is used, for the **grazing** incident case , the disperse of antenna in different frequency together with the reflection of ground will leads to the aliasing of the pulse and the arriving time based on peak value detection. while this is always the cased for beacon based calibration

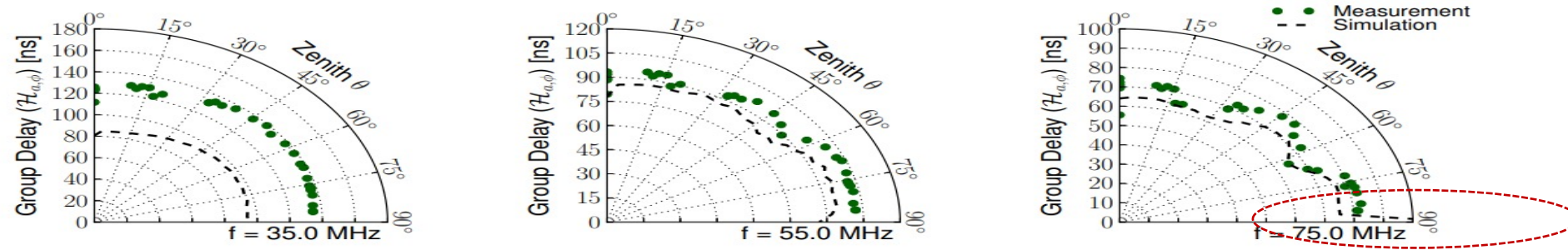


Figure 12: The group delay of the Small Black Spider LPDA as a function of zenith angle for three different frequencies in the measurements and the simulations.

See in : Antennas for the Detection of Radio Emission Pulses from Cosmic-Ray induced Air Showers at the Pierre Auger Observatory

- $T_{R\_RF\_Chain} + T_{Feb\_sample}$  present for the remain RF respond time and the Feb sample time , which always uniform and stable.

The time delay data during test include the GPS \_Time as recording reference:

$$T_{TR\_beacon}^{Test}(n) = T_{T\_transmitter} + T_{P\_process}(n) + T_{R\_antenna}(n, \theta, \varphi) + T_{R\_RF\_Chain} + T_{Feb\_sample} \\ + \sigma_{GPS\_SR}(n) + \sigma_{GPS\_LF}(n) + \sigma_{GPS\_Feb}(n)$$

The relative time delay difference:

$$T_{TR\_beacon}^{Test}(n) - T_{TR\_beacon}^{Test}(n_{ref}) = T_{P\_process}(n) + T_{R\_antenna}(n, \theta, \varphi) + \sigma_{GPS\_SR}(n, t) + \sigma_{GPS\_LF}(n, t) + \sigma_{GPS\_Feb}(n)$$

The calibrated parts:

$$\Delta T_n^{ref} = T_{TR\_beacon}^{Test}(n) - T_{TR\_beacon}^{Test}(n_{ref}) = \Delta l(n) / (c / n_{air}) + T_{R\_antenna}(n, \theta, \varphi) + \sigma_{GPS\_SR}(n, t) + \sigma_{GPS\_LF}(n, t) + \sigma_{GPS\_Feb}(n)$$

Based on a RTK test of the location of all units , this can be obtained with a 1cm error, and results in 0.03ns

This is a challenge parts, while we can using a repeat test of data as AERA done or a test based on cable cross check to obtain it at 0.5ns precision

This is the part that should be obtain during calibration



# Discuss

$$\Delta T_n^{ref} = T_{TR\_beacon}^{Test}(n) - T_{TR\_beacon}^{Test}(n_{ref}) = \Delta l(n) / (c / n_{air}) + T_{R\_antenna}(n, \theta, \varphi) + \sigma_{GPS\_SR}(n, t) + \sigma_{GPS\_LF}(n, t) + \sigma_{GPS\_Feb}(n)$$

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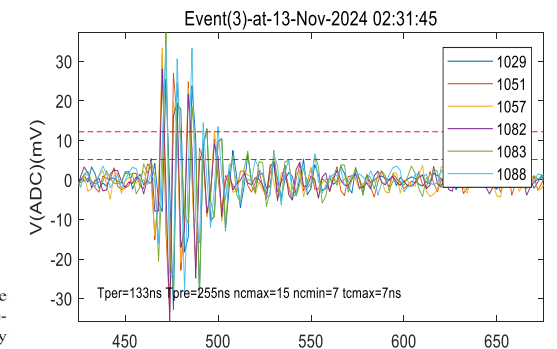
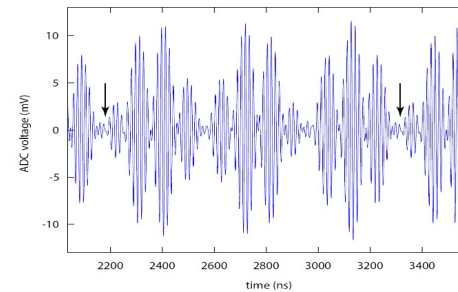
- $\sigma_{GPS\_SR}(n, t) + \sigma_{GPS\_LF}(n, t)$  ask for a real time calibration, which mean that the beacon signal will always in the spectrum data(As AERA), which lead limited influence to the event detection;
- $l(n) - l(ref)$  should be tested in the precision of 0.03m, if we hope to obtain 0.2ns precision in time domain;
- The remain question is how to obtain the  $T_{TR\_beacon}(n) - T_{TR\_beacon}(n_{ref})$
- This will based on the strategy of the choice of wave shape and the related trigger algorithm.

AERA using 4 sin wave and the related FFT phase information of  $\Delta\phi_j$  to obtain the

GP13 using pulse

Best time delay  $\tau$  :

$$\begin{aligned} \overline{EC}(\Delta\tau) &= \sum_j \int A_{1,j} A_{2,j} \sin(\omega_j t) \sin(\omega_j(t - \Delta\tau) + \Delta\phi_j) dt \\ &\propto \sum_j A_{1,j} A_{2,j} \cos(\Delta\phi_j - \omega_j \Delta\tau) \end{aligned}$$



**Figure 2.** Beacon beat: a voltage-time trace recorded with the analog-to-digital converters (ADCs) of the north-south channel of an AERA station is digitally filtered such that it contains only the four beacon frequencies. The signal shape of the beacon beat has a periodicity of approximately 1100 ns, as indicated by

# Calibration source and what we can do based on them?

## (1) Fixed Beacon (long running)

Time calibration: (How to separate the contribution:)

GPS +group-delay + firmware record

- (weather +cloudy + GPSantenna + GPS chain+GPS \_Chip)
- Antenna+RF Chain +AD;
- Firmware confirm within laboratory

Those part will be done in laboratory

## (2) Moving beacon on truck ( Co-plan calibration)

Time calibration; (2000second fix time?)

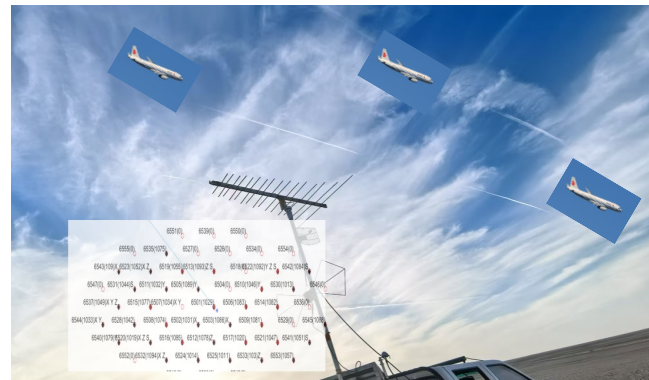
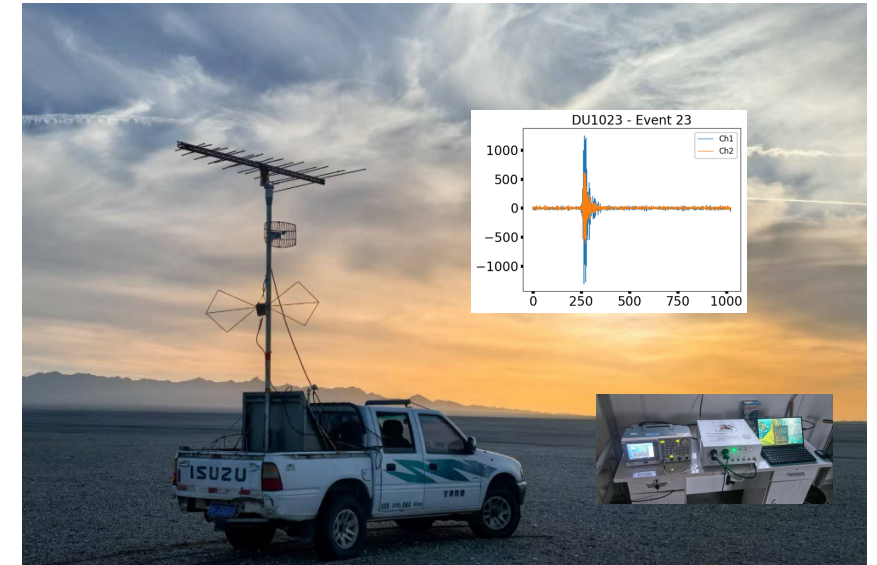
Location calibration;

## (3)Moving beacon in UAV(drone)

Pattern; Polarization(limitation)

## (4)Airplane

Pattern; Polarization?



# The first site test: Preparation for beacon test

Many thanks to: Yanhuang, Wang, xiao, Yizhang,  
Oliver, Palbolo, Fred, Bohao, Pengxiong

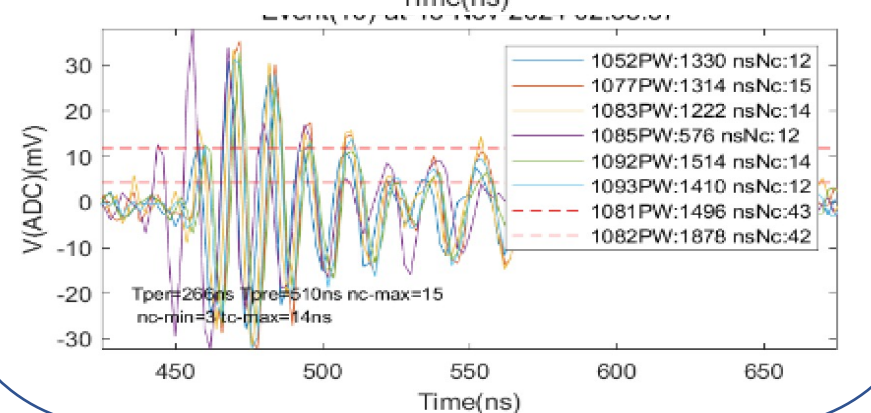
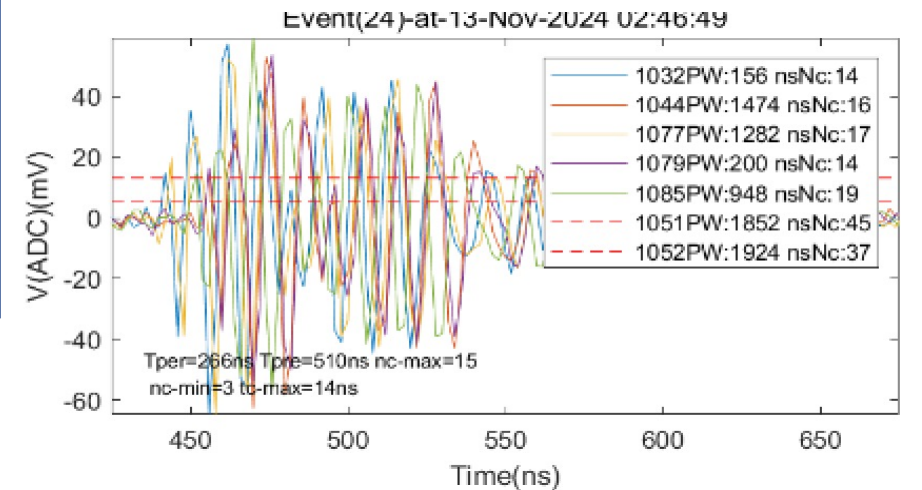
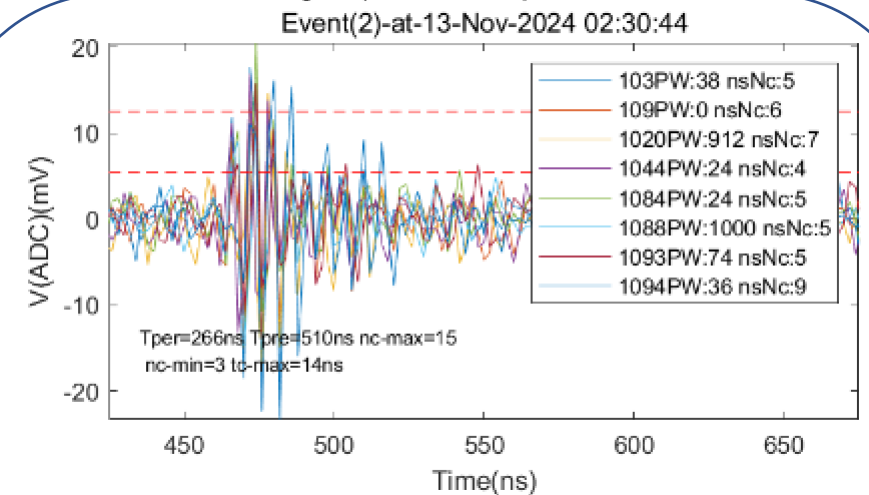
(1) Position information

(2) Beacon system hardware

- ◆ LPDA antenna
- ◆ PL
- ◆ Source
- ◆ Truck

(3) Trigger algorithm and pulse choice

(4) Gain/ radiation pattern calibration



# Statues of Units in GP13~ GP65

Mechanical + solar panel: 13

Charge controller: 13

Antenna: 13

LNA: 13,

Full function FEB: 11, Missing (2,3)

Now Work: 11(8)

Mechanical + solar panel: 65

Charge controller: 63 ; Missing: 119, 106

Antenna: 54; Missing 4 and others

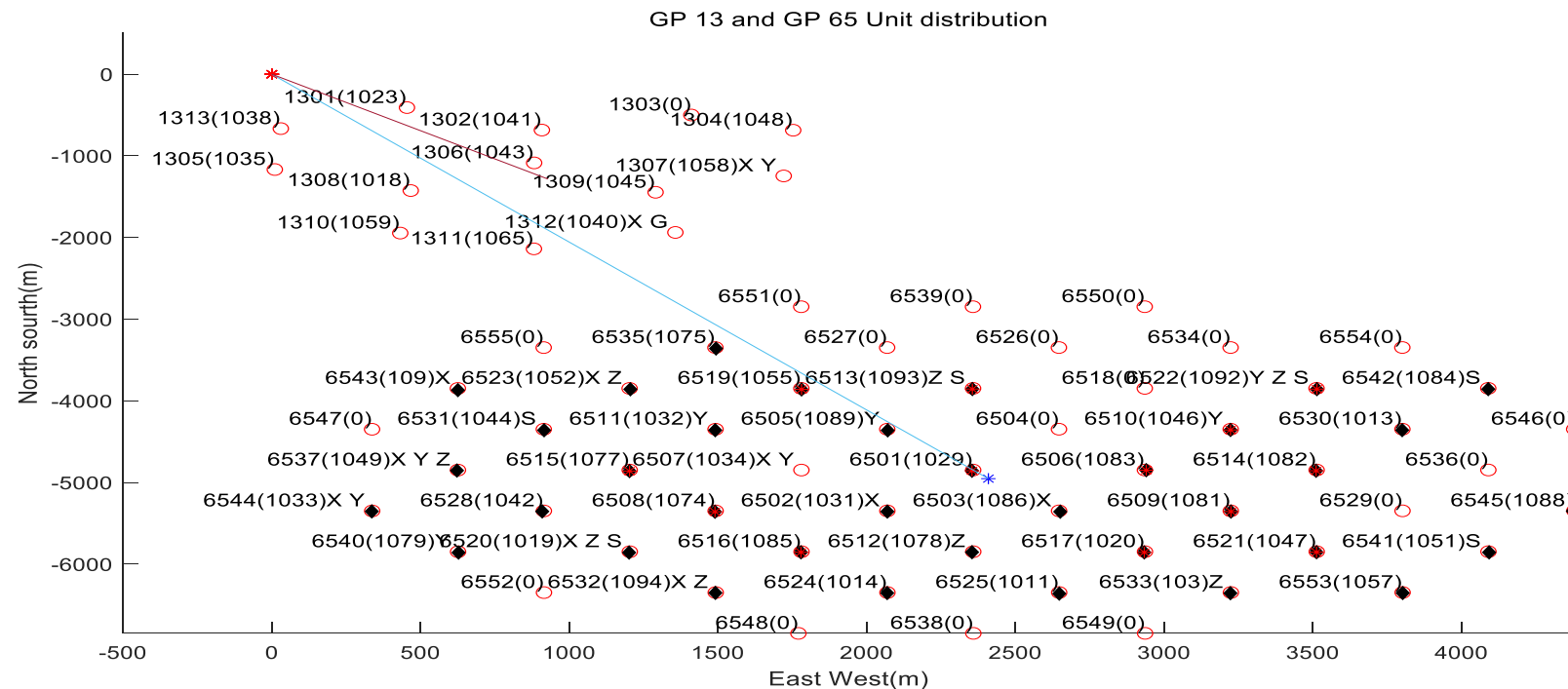
LNA: 54, not work: 36, 29, 26, 34, 54, 50, 18

Full function FEB: 38

Good statuses: 36, 12 (1078) ,33 (103) Z channel not good

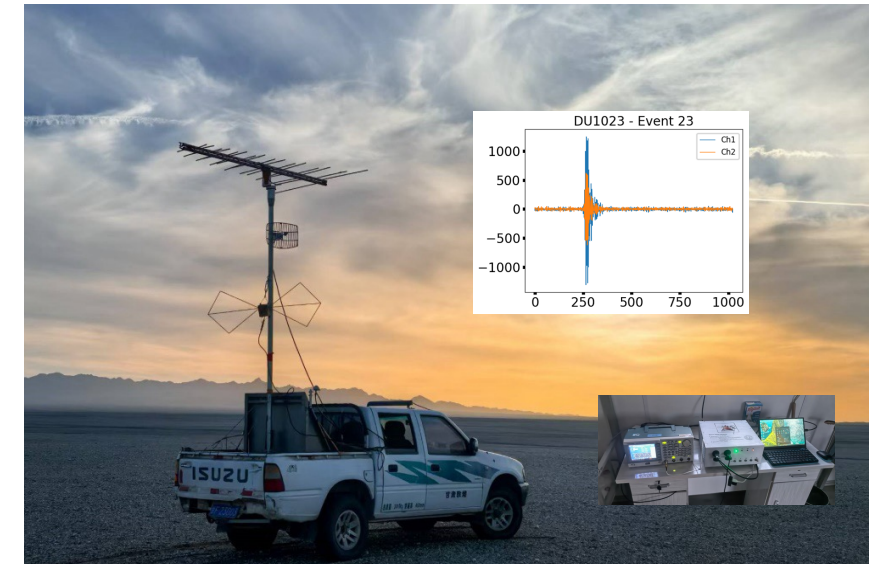
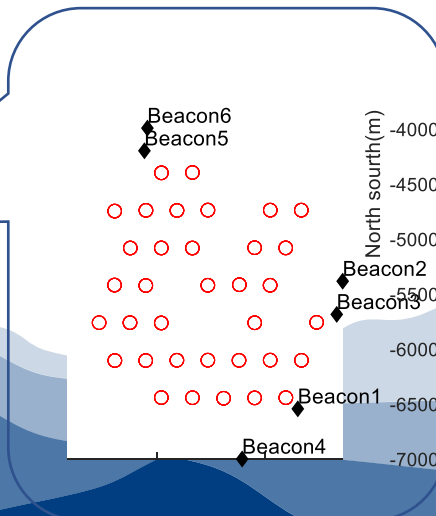
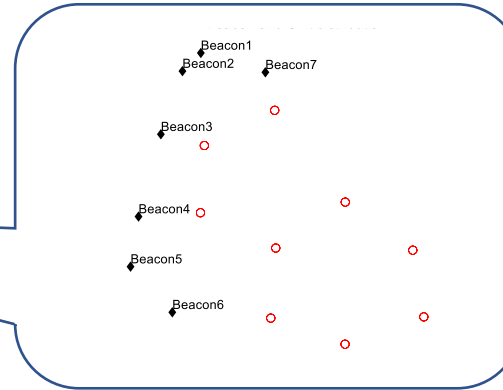
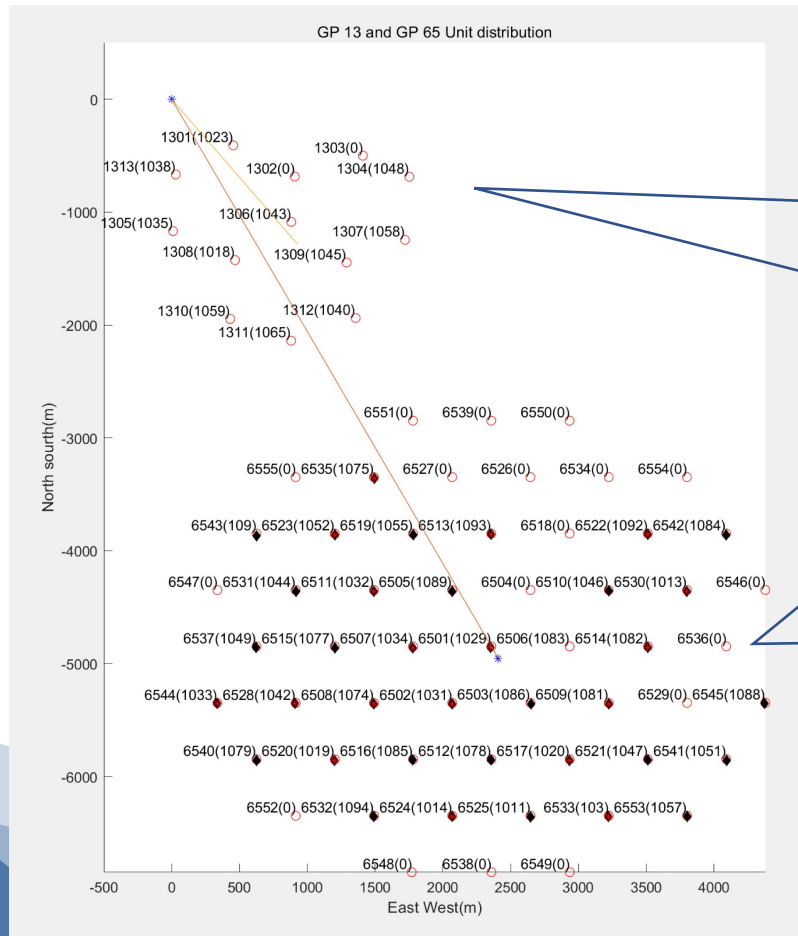
Now work: 35 (include 13&33) (Run 127:33Units)

Loss: 3 (1086) , 30 (1013) , 7 (1034)





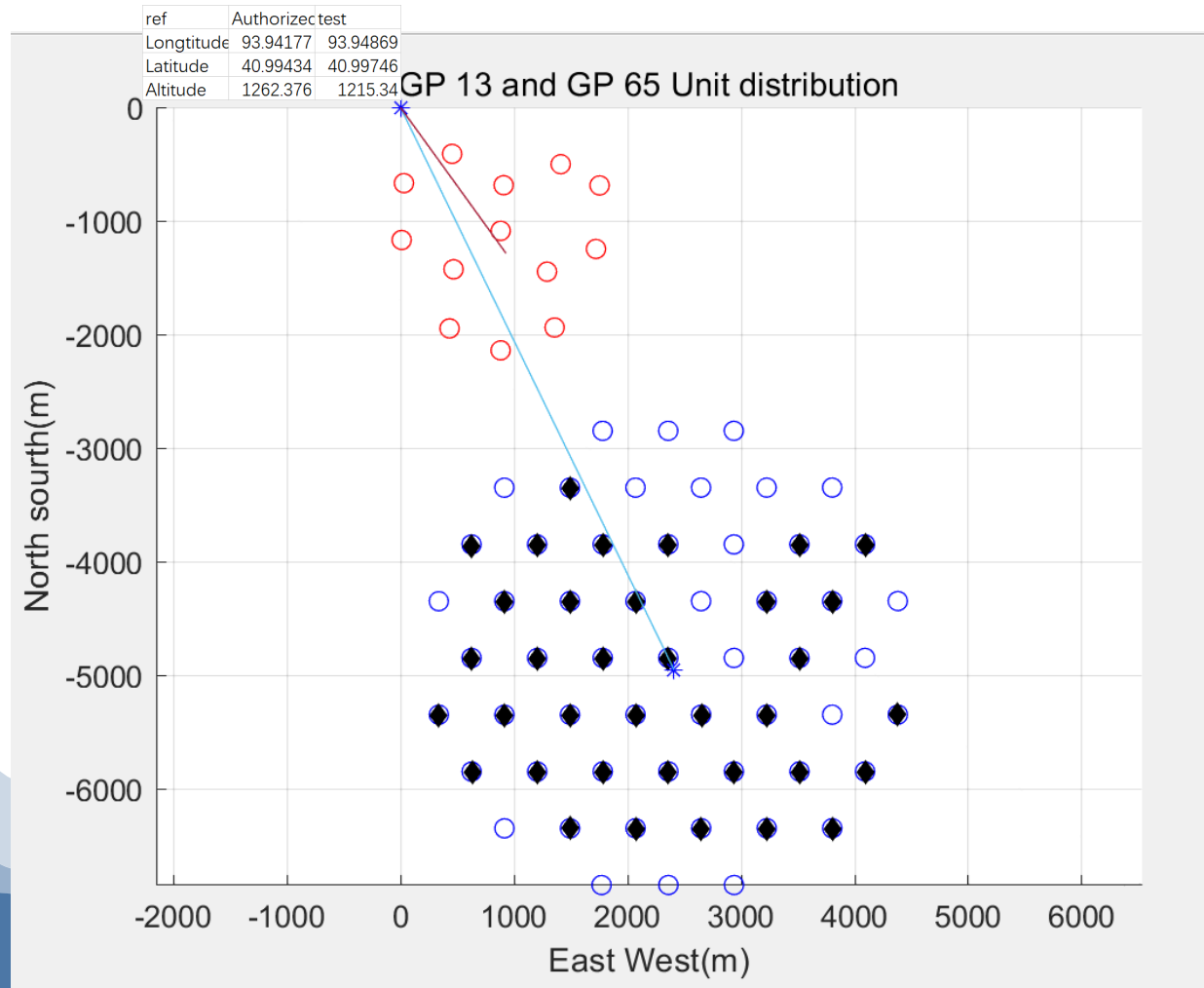
# GPS time calibration based on Beacon test



- Beacon  
(115MHz)  
3pulse  
100ms  
5V  
3minites

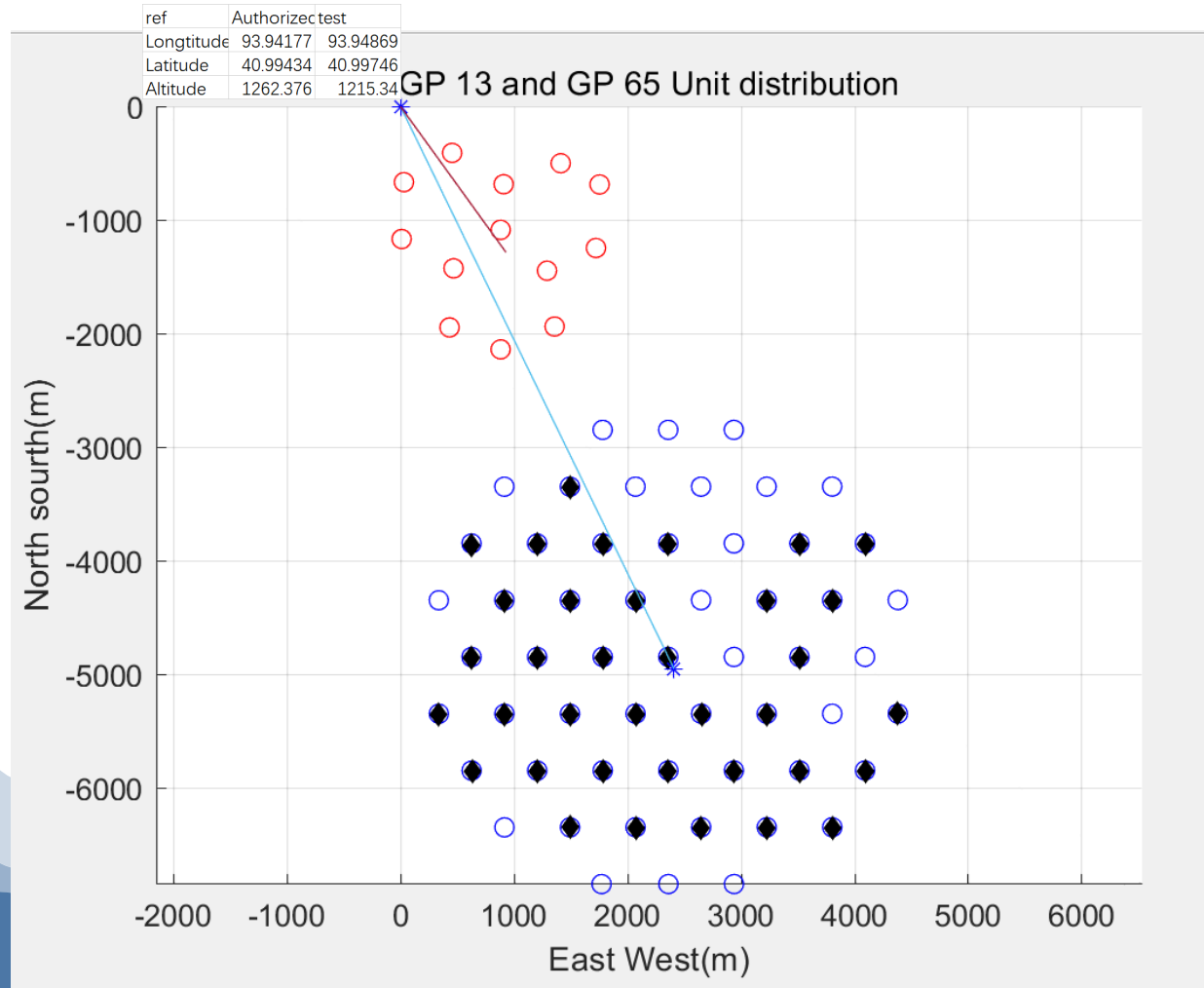
- DU  
1034+GP13/GP65  
No LNA  
L1 trigger  
X or Y trigger

# Statues confirm (GPS under test work well)



The starting GPS time of ID-103(Latitude=40.9402 Longitude=93.987 Altitude=1270.9638) is: 2024:10:23:22:2:20  
The starting GPS time of ID-109(Latitude=40.9626 Longitude=93.9561 Altitude=1234.7943) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1011(Latitude=40.9402 Longitude=93.9801 Altitude=1267.2851) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1013(Latitude=40.9582 Longitude=93.9938 Altitude=1228.6681) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1014(Latitude=40.9402 Longitude=93.9733 Altitude=1267.8271) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1019(Latitude=40.9447 Longitude=93.963 Altitude=1267.8574) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1020(Latitude=40.9447 Longitude=93.9835 Altitude=1263.2349) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1029(Latitude=40.9537 Longitude=93.9766 Altitude=1246.0239) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1031(Latitude=40.9492 Longitude=93.9733 Altitude=1255.5966) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1032(Latitude=40.9582 Longitude=93.9664 Altitude=1236.3247) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1033(Latitude=40.9493 Longitude=93.9527 Altitude=1261.3226) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1034(Latitude=40.9537 Longitude=93.9698 Altitude=1242.1395) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1042(Latitude=40.9492 Longitude=93.9595 Altitude=1260.9882) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1044(Latitude=40.9582 Longitude=93.9596 Altitude=1251.4855) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1046(Latitude=40.9582 Longitude=93.987 Altitude=1231.5985) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1047(Latitude=40.9447 Longitude=93.9904 Altitude=1259.4381) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1049(Latitude=40.9538 Longitude=93.9561 Altitude=1252.4754) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1051(Latitude=40.9447 Longitude=93.9973 Altitude=1268.3875) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1052(Latitude=40.9627 Longitude=93.963 Altitude=1249.7938) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1055(Latitude=40.9627 Longitude=93.9698 Altitude=1232.6568) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1057(Latitude=40.9402 Longitude=93.9938 Altitude=1269.9668) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1074(Latitude=40.9492 Longitude=93.9664 Altitude=1259.1177) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1075(Latitude=40.9672 Longitude=93.9664 Altitude=1231.4105) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1077(Latitude=40.9537 Longitude=93.963 Altitude=1249.3626) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1078(Latitude=40.9448 Longitude=93.9766 Altitude=1265.7503) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1079(Latitude=40.9447 Longitude=93.9561 Altitude=1275.3077) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1081(Latitude=40.9492 Longitude=93.987 Altitude=1254.6058) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1082(Latitude=40.9537 Longitude=93.9904 Altitude=1244.0918) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1084(Latitude=40.9627 Longitude=93.9973 Altitude=1220.9782) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1085(Latitude=40.9447 Longitude=93.9698 Altitude=1253.4985) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1086(Latitude=40.9492 Longitude=93.9802 Altitude=1247.3204) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1088(Latitude=40.9493 Longitude=94.0006 Altitude=1246.4776) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1089(Latitude=40.9582 Longitude=93.9733 Altitude=1235.9529) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1092(Latitude=40.9627 Longitude=93.9904 Altitude=1231.9989) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1093(Latitude=40.9627 Longitude=93.9767 Altitude=1228.4018) is: 2024:10:23:22:2:20  
The starting GPS time of ID-1094(Latitude=40.9403 Longitude=93.9664 Altitude=1268.7551) is: 2024:10:23:22:2:20

# Calibration of Time based on given GPS location



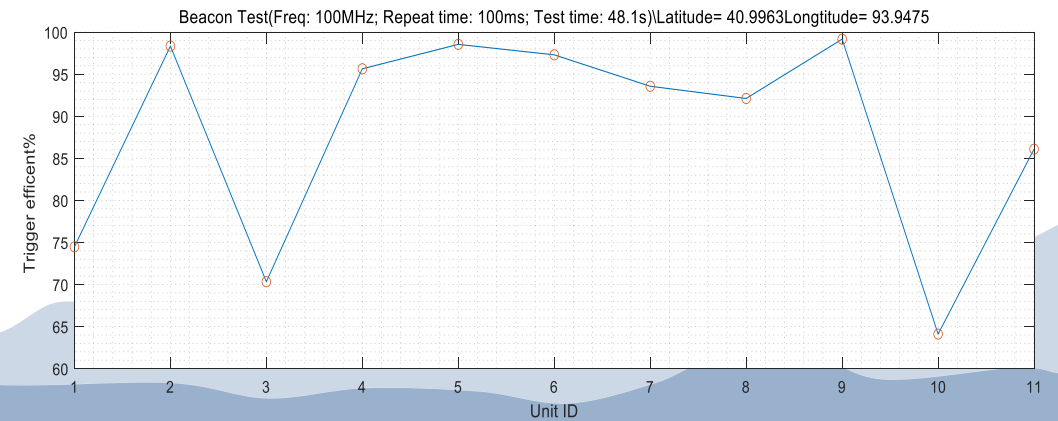
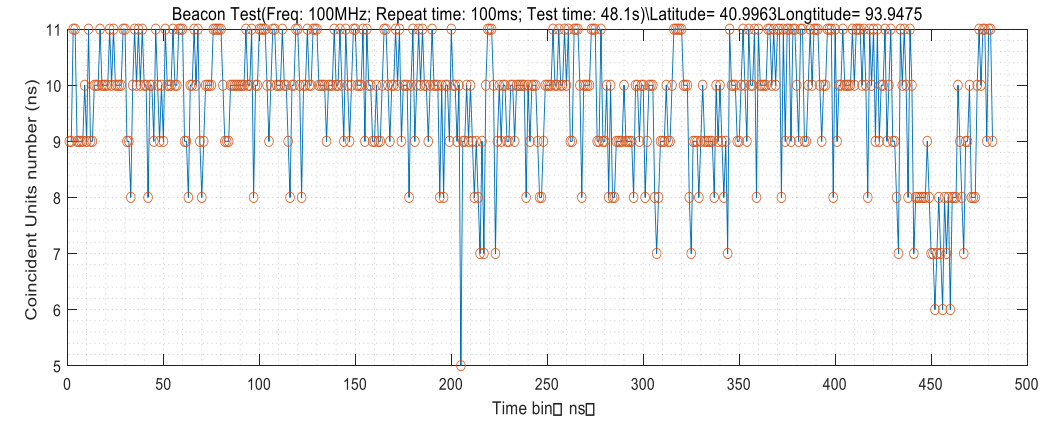
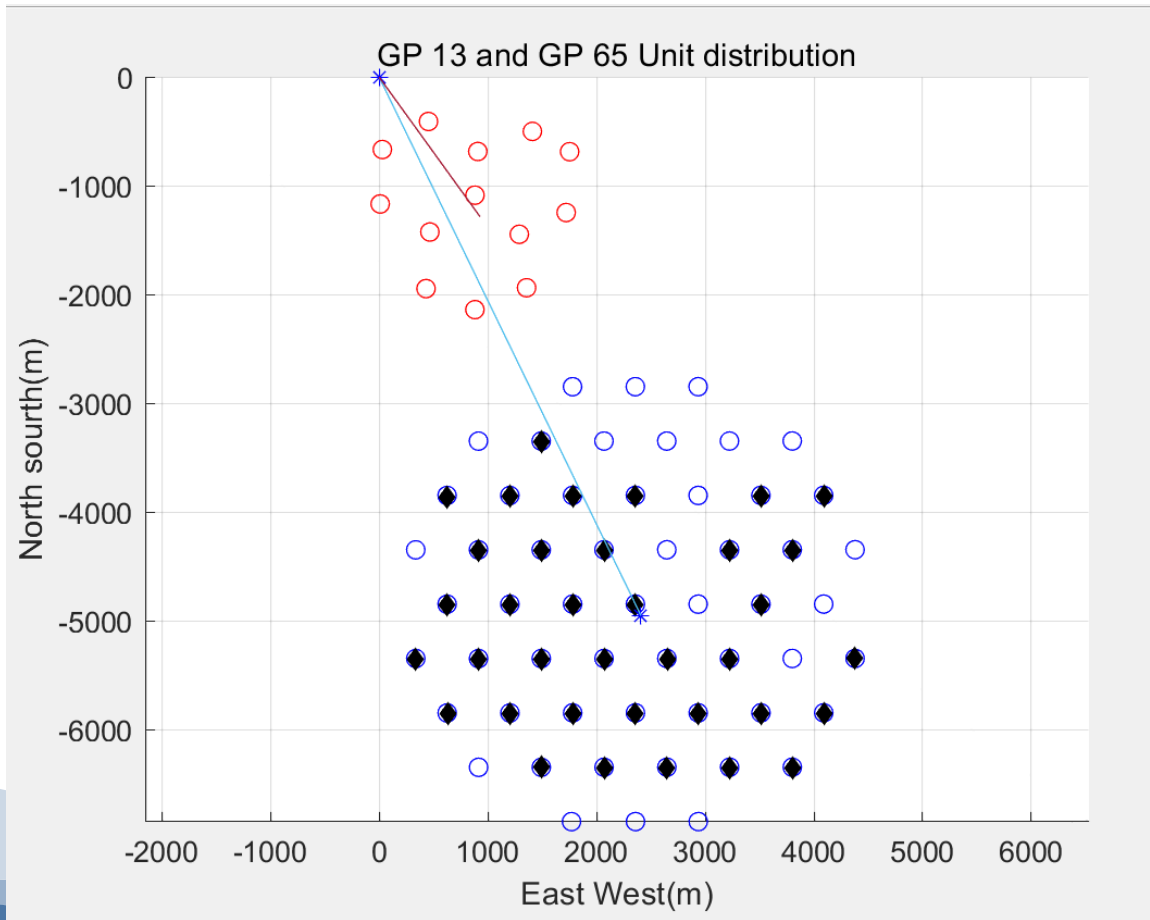
$$t_n^i(k) = t_{1034}^{\text{Beacon}} + \frac{l_n^i}{c} + \Delta t_n$$

$$t_n^i = \frac{1}{K} \sum_{k=1}^K t_n^i(k)$$

$$\Delta t_n = t_n^i - \left( t_{1034}^{\text{Beacon}} + \frac{l_n^i}{c} \right)$$

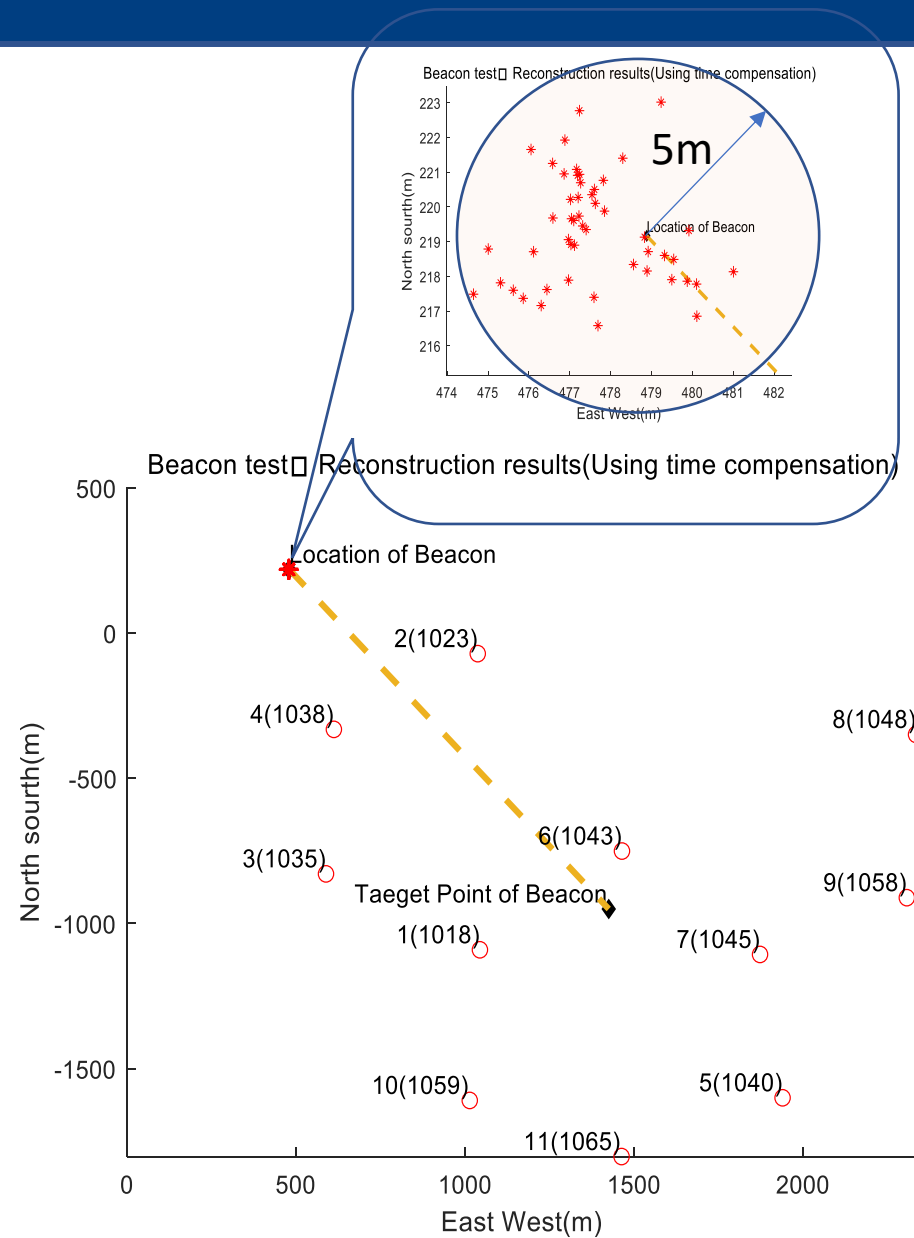
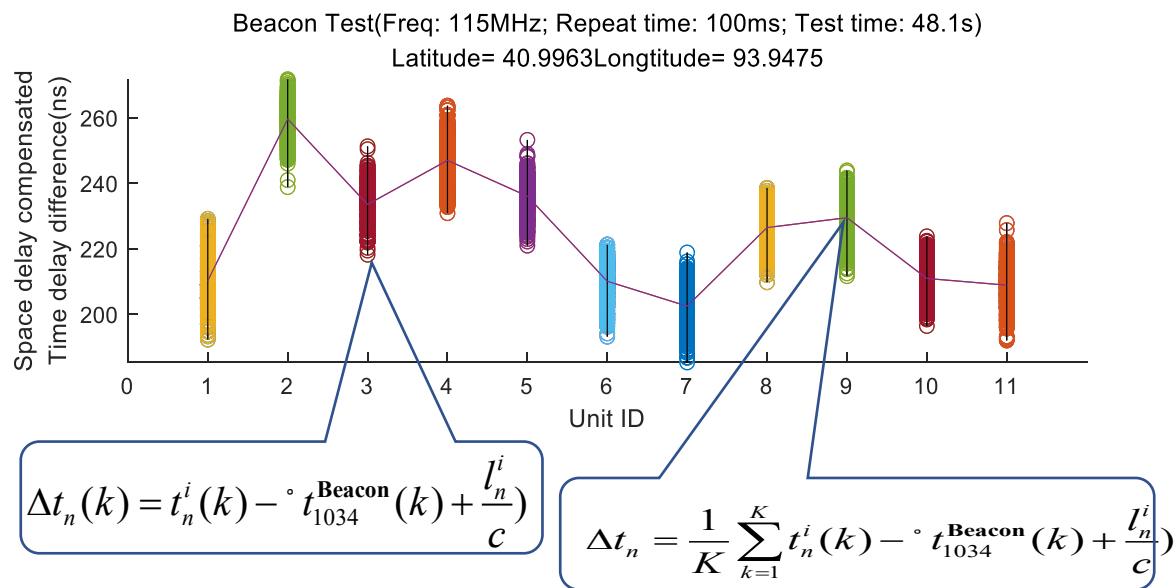
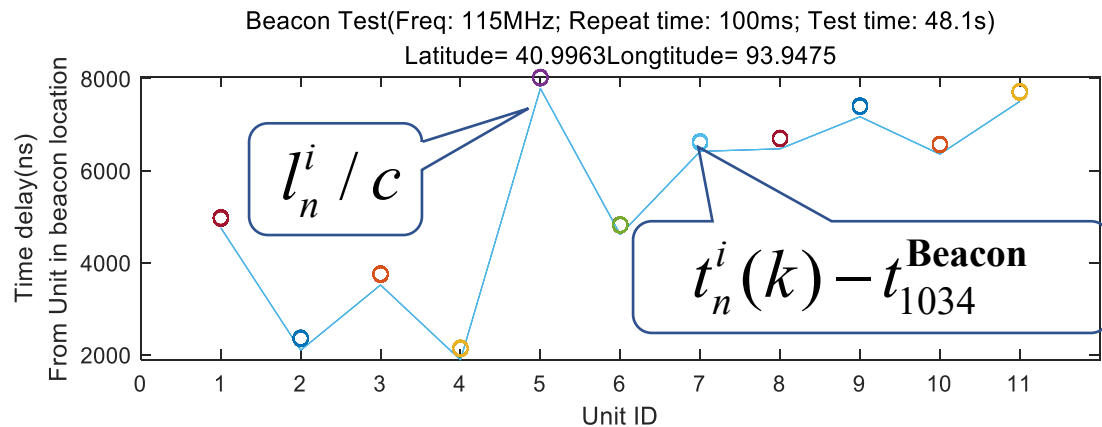
For each Beacon position  $i=1 \sim I$  ( $I=8$ , The number of different beacon locations), we send  $k$  pulse, and the collect the arriving time for all units (include the reference DU1034), the arriving time for all units  $n=1 \sim N$ , hence the time delay difference can be calibrate based on the average value in test.

# Data selection: time window=200ns, N\_trigger\_min=5

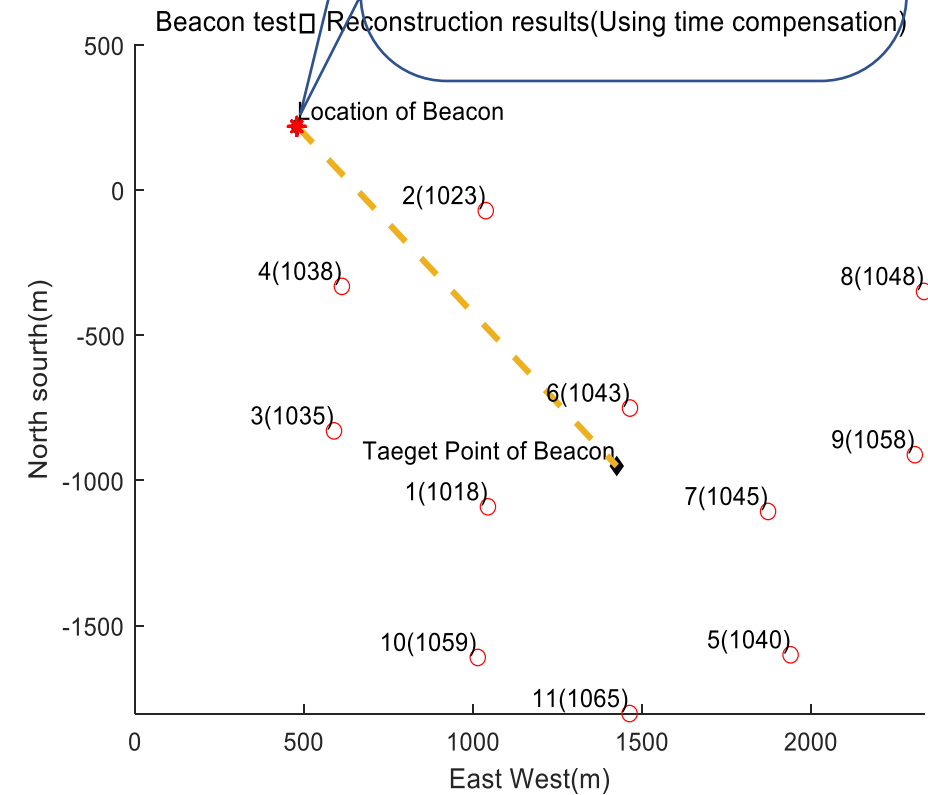
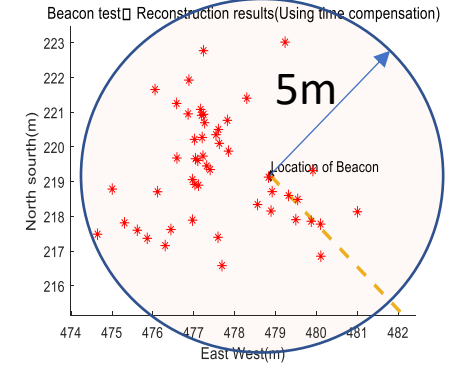
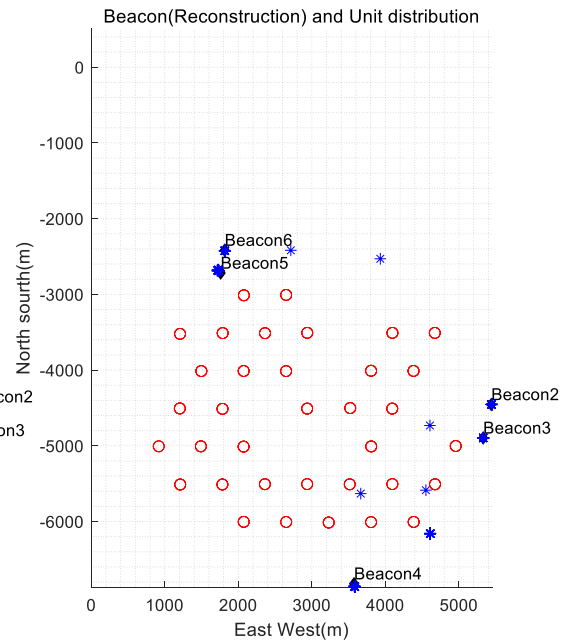
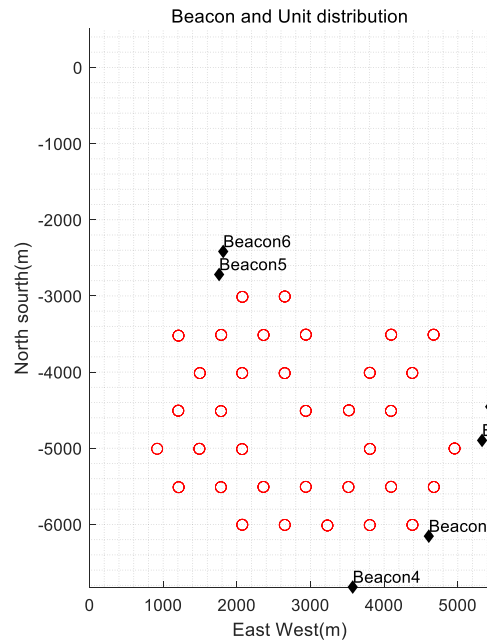
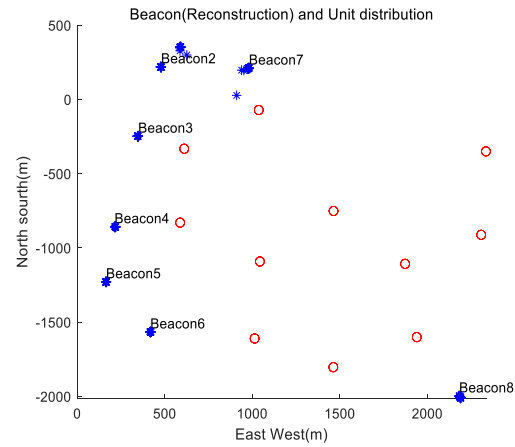
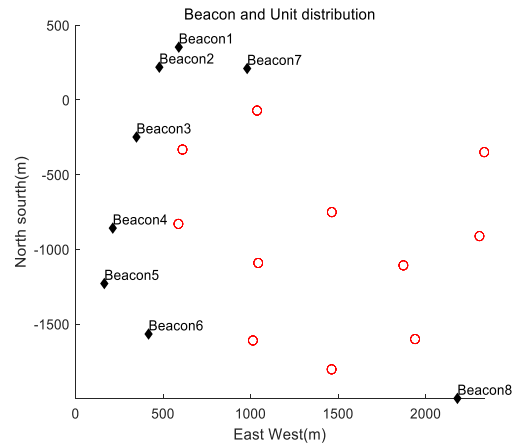




# Typical results

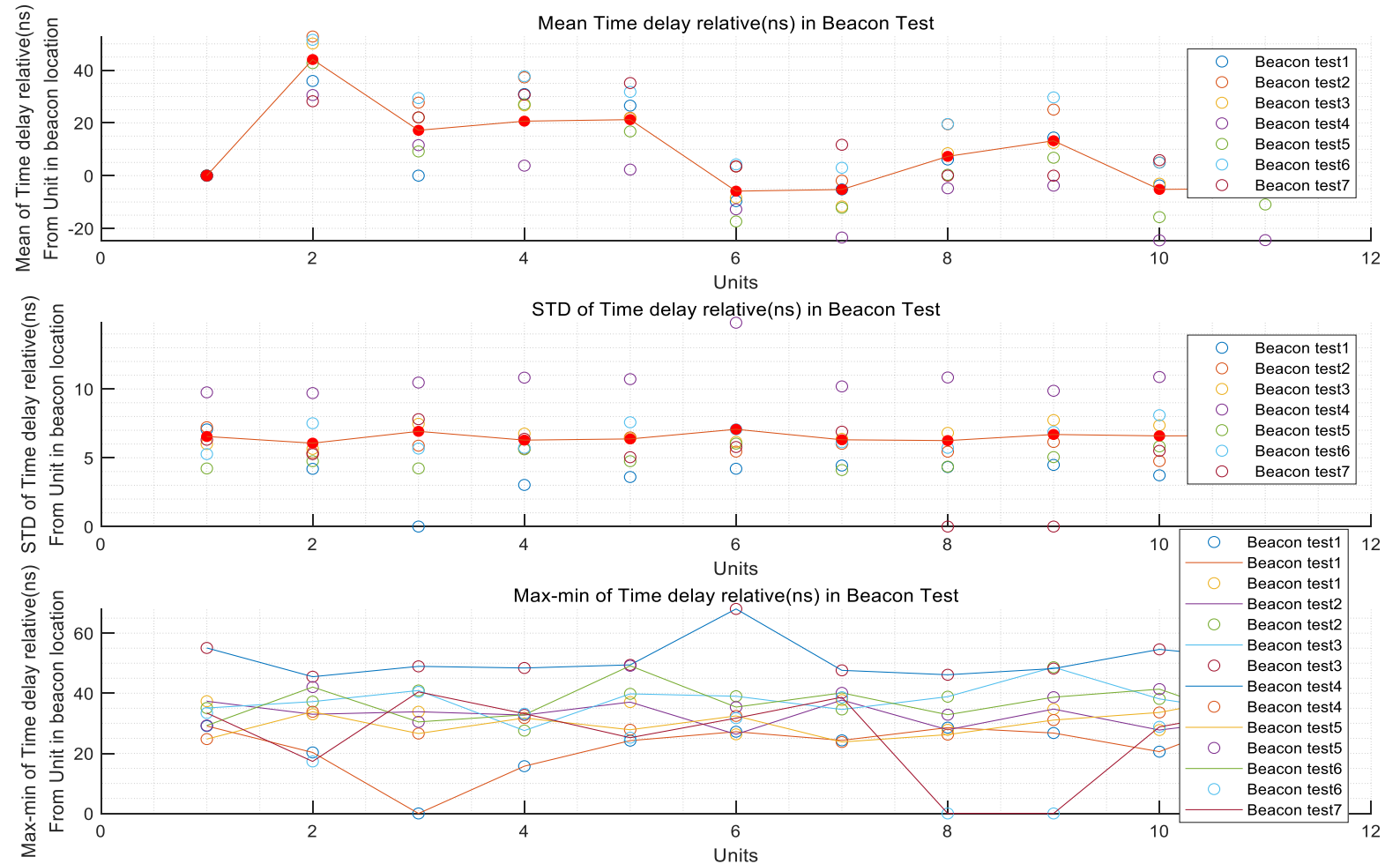
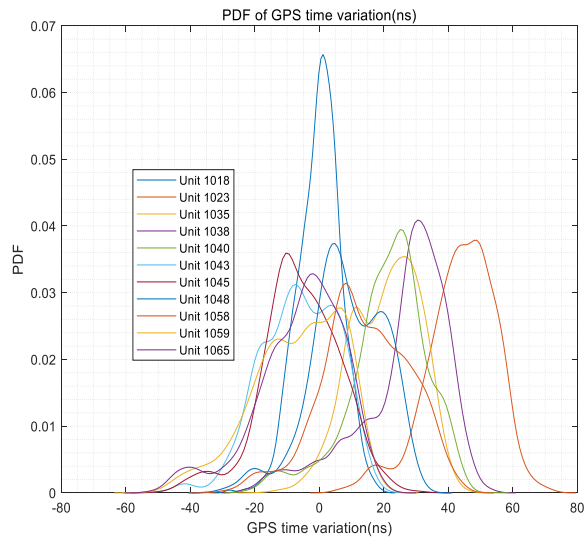


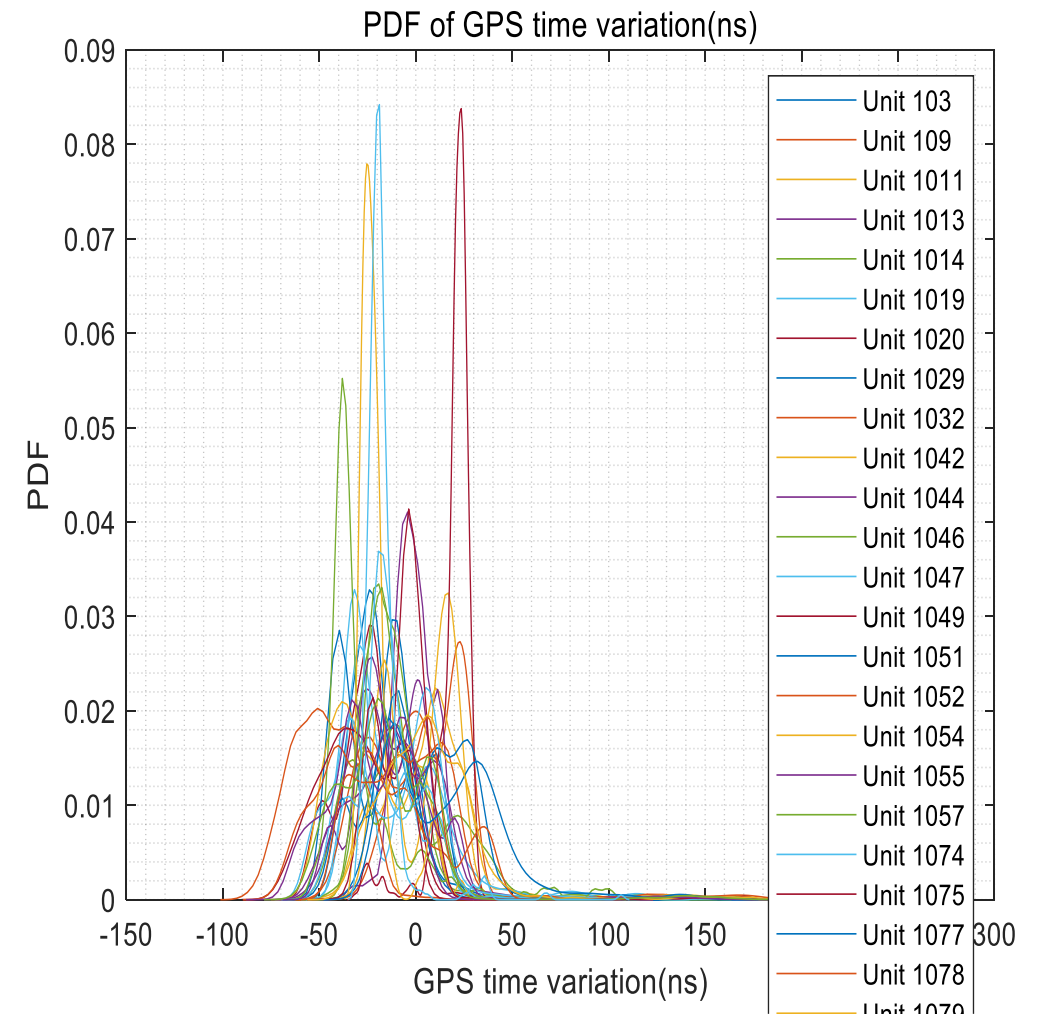
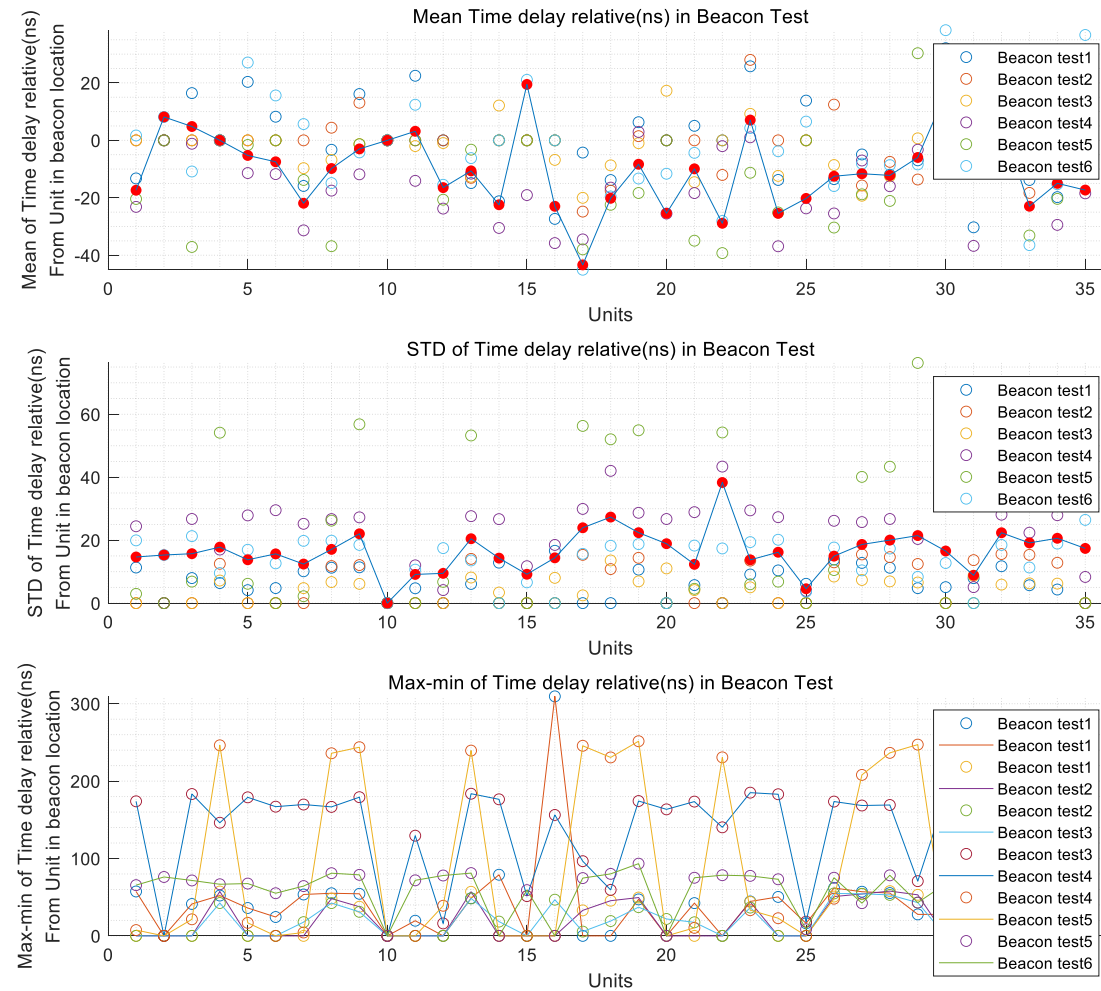
# Reconstruct results



# Muti\_beacon test results

- GPS time for different Units has a relatively stable difference of up to 50ns.(Represented by systematic differences in system response times)
- The 1 sigma accuracy of GPS is about 7ns;
- Some GPS time distribution is double peaks.





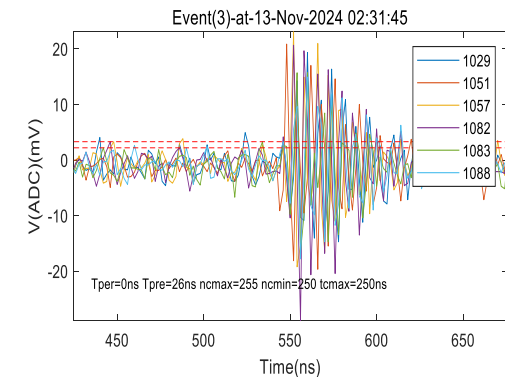
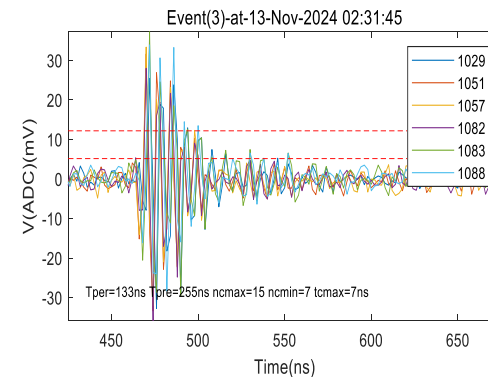
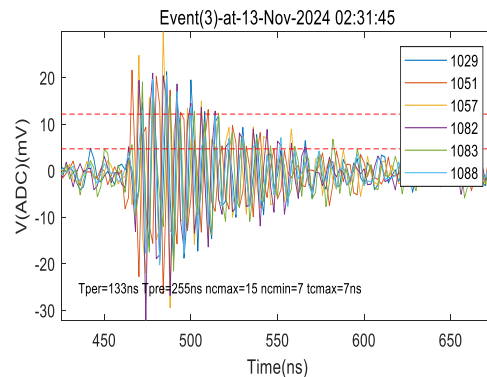
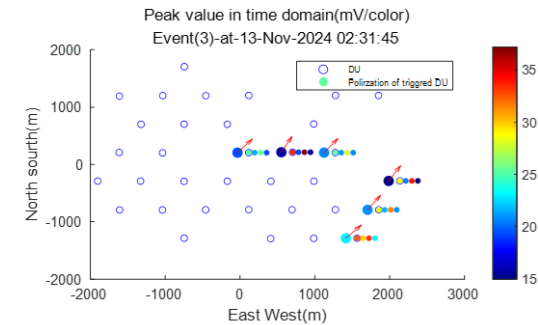
# Air plane-base calibration

Advantage of Airplane:

- ❑ (1) Free (almost for all time and All-weather work);
- ❑ (2) Moving (generate a trace that can used in different direction);
- ❑ (3) Relative stable (Magnitude\ spectrum and polarization);

Uncertainty of Airplane:

- Location(Direction, distance)
- Signal magnitude;
- Polarization
- Trace





# Another choice : Calibration Based on Wifi RF Chain

- 1: High frequency and time resolution
- 2: off-the-shelf hardware based on rocket and bullet
- 3: No ground reflection and

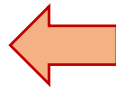
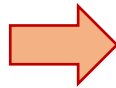
$T_{R\_antenna}$  is stable and uniform

Signal generator

200M  
up-conversion->5G

5G  
down-conversion->200M

Time calibration



## GRAND wifi system

Brand: UBIQUITI

Station: BULLET

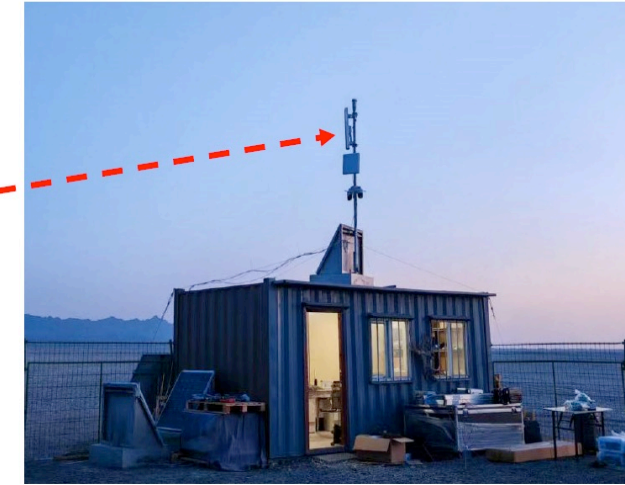
Access Point: ROCKET

Mode: PtMP

1 rocket: PtMP max. 330+ Mbps(Datasheet)

330 >> 50Mbps

PACKAGE LOSS IN DUNHUANG !!! Why?



Question :is it necessary to keep the GPS system for all units in next generation ?  
Or a Local crystal oscillator clock is enough.



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# Discuss

$$\Delta T_n^{ref} = T_{TR\_beacon}^{Test}(n) - T_{TR\_beacon}^{Test}(n_{ref}) = \Delta l(n) / (c / n_{air}) + T_{R\_antenna}(n, \theta, \varphi) + \sigma_{GPS\_SR}(n, t) + \sigma_{GPS\_LF}(n, t) + \sigma_{GPS\_Feb}(n)$$

Based on a RTK test of the location of all units , this can be obtained with a 1cm error, and results in 0.03ns

This is a challenge parts, while we can using a repeat test of data as AERA done or a test based on cable cross check to obtain it at 0.5ns precision

This is the part that should be obtain during calibration

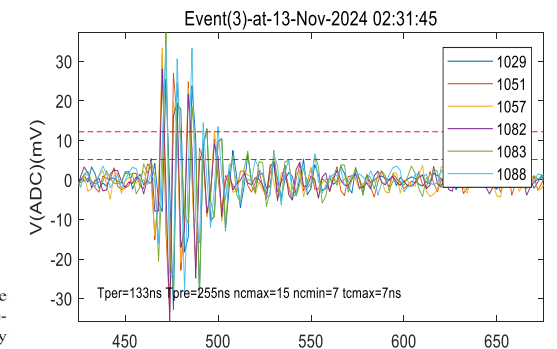
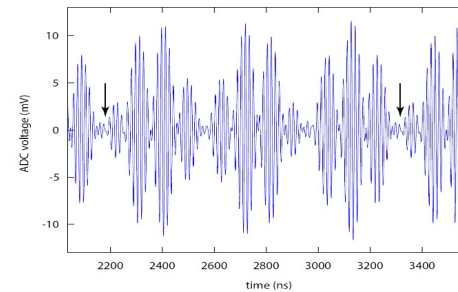
- $\sigma_{GPS\_SR}(n, t) + \sigma_{GPS\_LF}(n, t)$  ask for a real time calibration, which mean that the beacon signal will always in the spectrum data(As AERA), which lead limited influence to the event detection;
- $l(n) - l(ref)$  should be tested in the precision of 0.03m, if we hope to obtain 0.2ns precision in time domain;
- The remain question is how to obtain the  $T_{TR\_beacon}(n) - T_{TR\_beacon}(n_{ref})$
- This will based on the strategy of the choice of wave shape and the related trigger algorithm.

AERA using 4 sin wave and the related FFT phase information of  $\Delta\phi_j$  to obtain the

GP13 using pulse

Best time delay  $\tau$  :

$$\begin{aligned} \overline{EC}(\Delta\tau) &= \sum_j \int A_{1,j} A_{2,j} \sin(\omega_j t) \sin(\omega_j(t - \Delta\tau) + \Delta\phi_j) dt \\ &\propto \sum_j A_{1,j} A_{2,j} \cos(\Delta\phi_j - \omega_j \Delta\tau) \end{aligned}$$

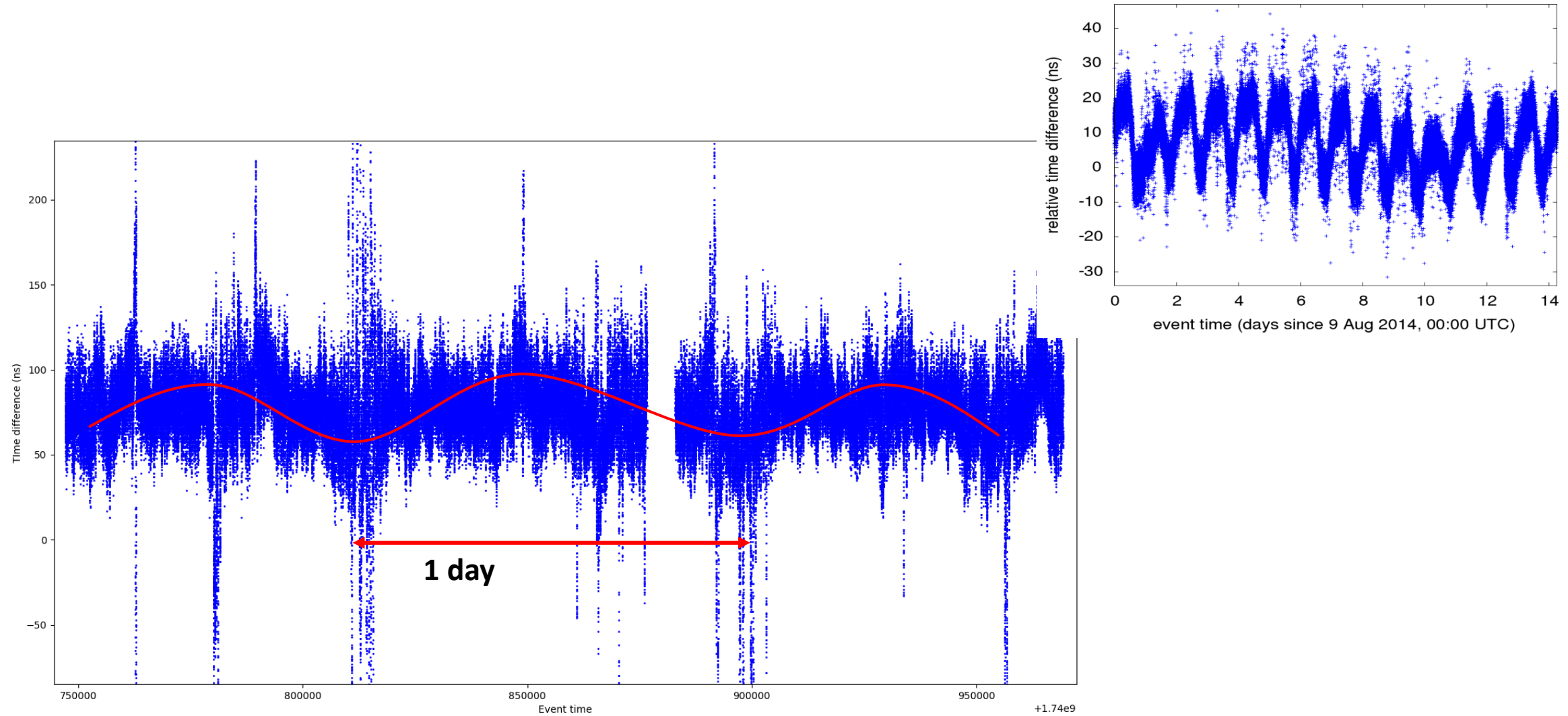


**Figure 2.** Beacon beat: a voltage-time trace recorded with the analog-to-digital converters (ADCs) of the north-south channel of an AERA station is digitally filtered such that it contains only the four beacon frequencies. The signal shape of the beacon beat has a periodicity of approximately 1100 ns, as indicated by

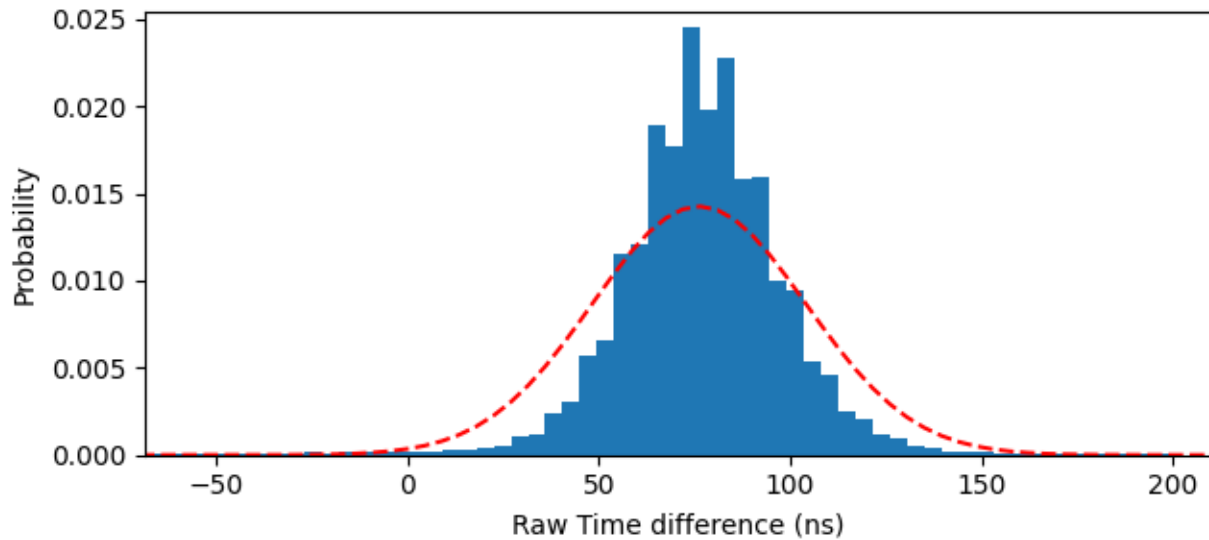
# Summary

- (1) We have figured out the factors that can caused error in GPS time, and we do the related calibration based on moving beacon in Dunhuang.
- (2) Calibration of position and antenna need improved based on RTK;
- (3) Calibration of time with 2 ns is very challenge;
- (4) Need cross check from other tests.

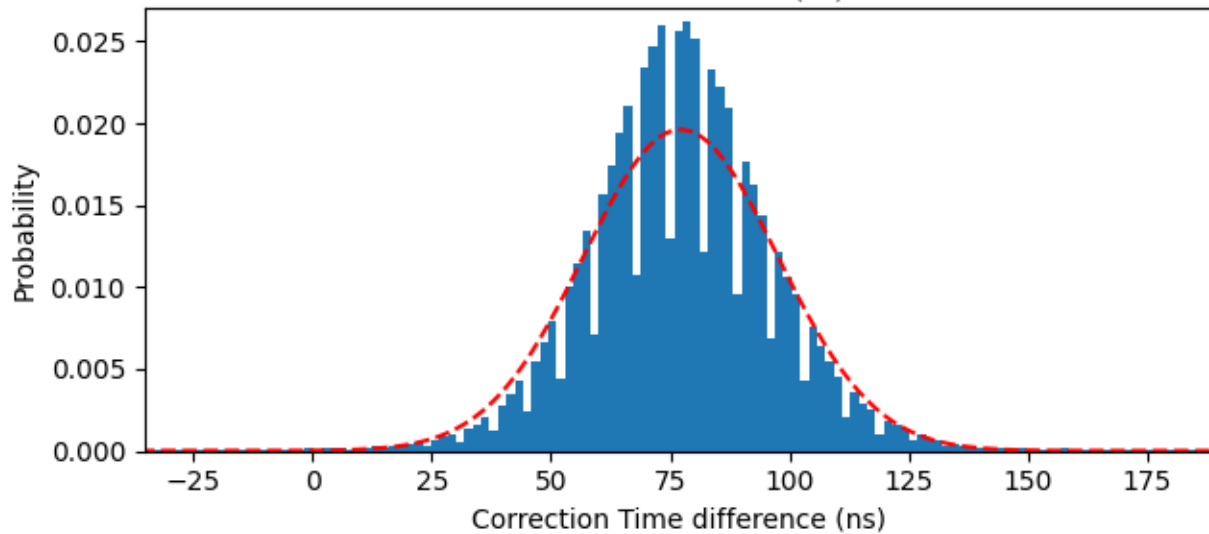
# (PMO Xingxu) Time Difference Periodical Change



# PMO Xing xu: Time Difference Distribution



$\sigma(\text{time diff}) = 27.98\text{ns}$   
 $\rightarrow \sigma(\text{single board}) = 19.78\text{ns}$



$\sigma(\text{time diff}) = 20.32\text{ns}$   
 $\rightarrow \sigma(\text{single board}) = 14.37\text{ns}$